# Business Airplanes 2014

# For now, there remains a chasm in demand between the long-range, large-cabin class and the rest of the turbofan market.

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usiness jet apartheid remained the dominant theme in 2013, as it has for the last five years since the world economy struggles to recover from its deepest downturn in eight decades. Most long-range, largecabin business aircraft manufacturers flourished while most light and midsize jet makers floundered. Total jet deliveries stabilized at 678, essentially in line with deliveries a decade ago, according to GAMA statistics.

Jet deliveries actually dipped about 1% from 2012 to 2013, but billing soared 23% with Gulfstream leading the way with 144 large-cabin deliveries. Bombardier shipped 62 Global 5000 and 6000 aircraft, Dassault Falcon Jet delivered 77 units and Embraer logged 21 Legacy 600/650 shipments.

Corporate profitability has more than doubled since the bottom of the recession. The S&P index is up 170% since first quarter 2009. For now, the DJIA seems solidly locked in above 16,000. So, large corporations, especially multinationals,

have fat profits to renew their large-cabin aircraft fleets.

Public companies aren't the only beneficiaries of the post-recession recovery. There now are more than 60,000 high net worth individuals (HNWIs) around the globe who have \$100 million or more in disposable assets, according to some surveys. The largest concentration of uberrich isn't in North American, it's in Asia.

Large corporations and top tier HNWIs, as a result, are fueling the sales of purpose-built business aircraft with \$30-million-plus price tags, ones that can fly 4,000+ nm. Top tier aircraft that can fly 5,000 to 7,000 nm and that sell for \$50- to \$100-million, are doing especially well. Bombardier, for instance, delivered 94% more Global 5000/6000 aircraft than shorter range Challenger 605 jets.

Gulfstream doesn't disclose delivery numbers for individual large-cabin aircraft, but it's well known that the G550 and G650 are faring much better than the G450, judging from relative resale prices.

Dassault shipped more Falcon 7X

trijets than all other Falcon models combined. Industry sources say that Dassault plans to announce another large cabin aircraft at EBACE 2014 in Geneva, capitalizing on the sales strength of Falcon 7X.

In the light-jet segment, it was a different, if not depressing, story. Textron Aviation's Cessna was hit particularly hard, delivering fewer Citations in 2013 than in any year since 1996. Deliveries of Bombardier Learjet 60XR continue to wind down as customers shift their interest toward the Learjet 85, which just made its first flight in April. The Canadian firm delivered 18 of its Learjet 70/75 aircraft, thoroughly revamped versions of Learjet 40XR/45XR, late in the fourth quarter. The late year, Learjet 70/75 delivery rush buoys prospects for a better 2014. But, company chairman Guy Hachey cautions that the overall "global economy has remained persistently sluggish," damping expectations for a full-blown business aircraft recovery in 2014.

Beechcraft, newly merged into Textron Aviation, fared even more poorly with its turbofan aircraft. It ceased production of all jets and disposed of its remaining Hawker 4000 aircraft at fire sale prices.

Deep discounting by U.S. light jet makers remained a dominant practice in 2013. The downside of new aircraft discount-ing, though, is pronounced price softness in the used light-iet market Many older in the used light-jet market. Many older

light jets have so little residual value that operators are facing stiff cash outlays when trading up to new aircraft. That's another drag on new light jet sales.

Undeterred, Eclipse Aerospace is pressing ahead with its Eclipse 550 very light jet, having a \$2.85 million base price. Eclipse is betting that the upgraded Eclipse 500 will hold its own in the market because of its rock bottom operating costs.

Equally optimistic is Honda Aircraft Company, proceeding amain with development of its \$4.5 million, twin-turbofan HA-420 HondaJet. It's a direct competitor with Embraer Phenom 100 and Cessna Citation M2, the upgraded version of CJ1+.

Bombardier is pressing ahead at full speed with Learjet 85, the Canadian firm's new midsize jet that made its first flight in mid-April. The entry-into-service data for the all-composite, transcontinental U.S. range jet has yet to be determined, but it should reach full production rates by 2018, according to RBC Europe's market analyst Robert Stallard.

At present, Embraer remains in a strong position with its Phenom light jets. Last year, it delivered 60 Phenom 300 light jets, grabbing market share mainly at the expense of Cessna Citation CJ3 and CJ4. Admittedly, the Brazilian firm saturated the entry level jet segment in 2009 and 2010 when it delivered a total of 197 Phenom 100 aircraft. But, it still delivered more Phenom 300 units in 2013 than either Citation Mustang or M2.

Cessna is fighting back against Phenom 300 with upgraded versions of its existing models, including the CJ3+ that is making its *Purchase Planning Handbook* debut this year.

Some in the industry, though, say that Cessna's historic reliance on derivative designs, many of which have their roots in the original 1969 FanJet 500, is leaving the door open for Embraer to enter with its clean sheet designs, such as the Phenoms, and also for planemaker HondaJet. Buyers are no longer satisfied with evolutionary Mr. Potato Head derivatives, distinguished mainly by changes in their plug-in body parts.

Now, Embraer is again introducing disruptive technology with its Legacy 500, the first fly-by-wire midsize business jet. (See our flight test report on page 62 in this issue). It's priced \$2 million higher than Citation Sovereign, but it's a cleansheet design with a much larger cabin having a flat floor. It has higher cruise speeds and more tanks-full payload.

With 3,000 nm range and a \$20 million

price tag, Legacy 500 also will compete with the midsize Learjet 85. Its supermidsize cabin even makes competitive with Bombardier Challenger 300, the bestselling super-midsize aircraft. Bombardier is countering the Brazilians by offering Challenger 350, a longer range, more capable, more fuel efficient version of Challenger 300.

Potentially dealing another one-two body punch to Cessna, similar to Phenom 100 and 300, Embraer's Legacy 450 is slated to enter service in 2015. It will compete head-to-head against Citation Latitude, a larger fuselage version of Citation Sovereign. Both aircraft have 2,500 nm range and similar price tags, but Legacy 450 has fly-by-wire flight controls, higher cruise speeds and a larger cabin.

The Legacy 450 is priced \$2 million above the Learjet 60XR, but it does virtually everything better than the aging Bombardier midsize jet, having a considerably larger cabin, more range and more tanks-full payload.

The growth in turboprop shipments was a boon to manufacturers. Deliveries grew more than 10% in 2013, according to GAMA. Leading the way was Textron Aviation's Beechcraft unit that experienced a 50+ surge in King Air shipments. More than half were King Air 350 models, many of which were delivered to Wheels Up, a "members only" firm that placed the largest turboprop order in history.

Piaggio Aero remains a notable exception to the success in the twin turboprop segment. Shipments sagged to just two new Avanti II aircraft last year as the firm reels from the crash of fractional ownership firm Avantair, leaving behind dozens of unairworthy Avanti aircraft because of shoddy maintenance. Many former Avantair aircraft now face cannibalization. (See "The Avantair Failure, Part 2" on page 30 in this issue.)

The single-engine turboprop segment was a bright spot last year. Deliveries of Cessna's 208 Caravan and 208B Grand Caravan reached 105 units, only two down from 2012 and higher than the annual average of the last two decades. Quest delivered 28 Kodiak 100 utility aircraft, the larger number in its history.

Deliveries of pressurized singles remain robust. Piper shipped 34 Meridian aircraft, Socata delivered 40 TBM 850 G1000 turboprops and Pilatus shipped 65 PC-12s. Socata's 330 KTAS TBM 900 is appearing for the first time in this year's Handbook. It has 21% shorter takeoff distances, superior time to climb and 17-kt. faster cruise speeds than TBM 850, along with slower stall speeds, more docile low speed handling and a quieter cabin. It's actually faster than some light jets on trips up to two hours duration.

Asking prices for most new single-engine turboprops, as a result, are firm. There's little motivation for most manufacturers, particularly Pilatus and Socata, to negotiate on list price.

Looking ahead at the remainder of 2014, small businesses in the U.S., firms that historically have purchased the majority of light jets, continue to struggle. Owners remain unsettled about the prospects for a broad-based economic recovery, as well as the threat of new federal mandates. Most historic light jet buyers are in no mood to purchase new aircraft, keeping a tight rein on purse strings, preserving capital for unknown threats ahead.

The Federal Reserve has expressed concerns that U.S. economic inflation is too low at 1%, half the target rate for healthy economic growth. Low consumer prices would seem to be a boon, but low inflation also signifies poor wage growth, high unemployment and excess economic capacity. European economic inflation is near zero, prompting concerns that the world economy could be at risk for deflation. Such news rattles the confidence of small businesses.

Moreover, the pre-owned light jet market remains awash with great deals to be scooped up by savvy shoppers. True, a certain segment of buyers only purchases new aircraft. For them, only a new Mustang, M2 or Eclipse 550, HondaJet or Learjet 75, will suffice.

But, large numbers of prospects, outside of that select few that only buy new, find themselves tempted by \$1.5 million Mustangs, \$1.9 million CJ1+ aircraft and Phenom 100s, and \$3.5 million Learjet 45XR aircraft, among other bargains in the basement.

So, for now, there remains a huge chasm in demand between the longrange, large-cabin class and the rest of the turbofan market. If you're shopping for a new aircraft with less than 5,000 nm of range, you'll find sales staffs willing to sharpen their pencils to get you to sign their purchase contracts. This year will be another bonanza for many buyers. **B&CA** 

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**B&CA's** digital edition contains Used Airplanes and Regional Aircraft comparative tables. The Purchase Planning Handbook is available for download at AviationWeek.com/bca

# How to Use the Airplane Charts

or an aircraft to be listed in the *Purchase Planning Handbook*, a production conforming article must have flown by May 1 of this year. The dimensions, weights and performance characteristics of each model listed are representative of the current production aircraft being built or for which a type certificate application has been filed. The Basic Operating Weights we publish should be representative of actual production turboprop and turbofan aircraft because we ask manufacturers to supply us with the average weights of the last 10 commercial aircraft that have been delivered. However, spot checks of some manufacturers' BOW numbers reveal anomalies. Prospective buyers are advised to verify the actual weights of aircraft with options.

The takeoff field length distances are based on Maximum Takeoff Weight unless otherwise indicated in the tables.

Please note that "all data preliminary" in the remarks section indicates that actual aircraft weight, dimension and performance numbers may vary considerably after the model is certified and delivery of completed aircraft begins.

### Manufacturer, Model and Type Designation

In some cases, the airplane manufacturer's name is abbreviated, but the company's full name and address can be found in the "Airframe Suppliers Directory" at our website. The model name and the type designation also are included in this group.

### **B&CA Equipped Price**

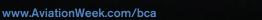
Price estimates are first quarter, current vear dollars for the next available delivery. Some aircraft have long lead times, thus the actual price will be higher than our published price. Note well, manufacturers may adjust prices without notification.

▶ Piston-powered airplanes – Computed retail price with at least the level of equipment specified in the B&CA Required Equipment List.

Turbine-powered airplanes – Average price of 10 of the last 12 commercial deliveries, if available. Some manufacturers decline to provide us with actual prices of delivered aircraft. The aircraft serial numbers aren't necessarily consecutive because of variations in completion  $\overline{2}$ 

N7321





time and because some aircraft may be configured for non-commercial, special missions.

### **Characteristics**

**Seating** – Crew + Typical Executive Seating/Maximum Seating.

For example, 2+8/19 indicates that the aircraft requires two pilots, there are eight seats in the typical executive configuration and the aircraft is certified for up 19 to passenger seats. A four-place single-engine aircraft is shown as 1+3/3, indicating that one pilot is required and there are three other seats available for passengers. We require two pilots for all turbofan airplanes, except for single-pilot certified aircraft such as the Eclipse 550, Cessna Citation CJ series and Syberjet SJ30-2, which have, or will have, a large percentage of single-pilot operators. Four crewmembers are specified for ultralong-range aircraft — three pilots and one flight attendant.

Each occupant of a turbine-powered airplane is assumed to weigh 200 lb., thus allowing for stowed luggage and carry-on items. In the case of piston-engine airplanes, we assume each occupant weighs 170 lb. There is no luggage allowance for piston-engine airplanes.

▶ Wing Loading – MTOW divided by total wing area.

Power Loading – MTOW divided by total rated horsepower or total rated thrust.
 FAR Part 36 certified noise levels – Flyover noise in A-weighted decibels (dBA) for small and turboprop aircraft. For turbofan-powered aircraft, we provide

Part 36 EPNdB (effective perceived noise levels) for takeoff, sideline and approach.

### Dimensions

**External length**, height and span dimensions are provided for use in determining hangar and/or tie-down space requirements.

Internal length, height and width are based on a completed interior, including insulation, upholstery, carpet, carpet padding and fixtures. Note well: These dimensions are not based upon metal-tometal measurements. They must reflect the actual net dimensions with all soft goods installed. Some manufacturers provide optimistic measurements, thus prospective buyers are advised to measure aircraft themselves.

As shown in the Cabin Interior Dimensions illustration, for small airplanes other than "cabin-class" models, the length is measured from the forward bulkhead ahead of the rudder pedals to the back of the rearmost passenger seat in its normal, upright position.

For so-called cabin-class and larger aircraft, we show the overall length of the passenger cabin, measured from the aft side of the forward cabin divider to the aft-most bulkhead of the cabin. The aft-most point is defined by the rear side of a baggage compartment that is accessible to passengers in flight or the aft pressure bulkhead. The overall length is reduced by the length of any permanent mounted system or structure that is installed in the fuselage ahead of the aft bulkhead. For example, some aircraft have full fuselage cross-section fuel tanks mounted ahead of the aft pressure bulkhead.

The second length number is the net length of the cabin that may be occupied by passengers. It's measured from the aft side of the forward cabin divider to an aft point defined by the rear of the cabin floor capable of supporting passenger seats, the rear wall of an aft galley or lavatory, an auxiliary pressure bulkhead or the front wall of the pressurized baggage compartment. Some aircraft have the same net and overall interior length because the manufacturer offers at least one interior configuration with the aft-most passenger seat located next to the front wall of the aft luggage compartment.

Interior height is measured at the center of the cross section. It may be based on an aisle that is dropped several inches below the main cabin floor that supports the passenger seats. Some aircraft have dropped aisles of varying depths, resulting in less available interior height in certain sections of the cabin, such as the floor sections below the passenger seats.

Two width dimensions are shown for multiengine turbine airplanes — one at the widest part of the cabin and the other at floor level. The dimensions, however, are not completely indicative of the usable space in a specific aircraft because of individual variances in interior furnishings.

### Power

▶ Number of engines, if greater than one, and the abbreviated name of the manufacturer: Honeywell, CFMI — CFM International, TCM — Teledyne Continental, IAE — International Aero Engines, Lyc — Textron Lycoming, P&WC — Pratt & Whitney Canada, RR — Rolls-Royce and Wms — Williams International.

▶ Output – Takeoff rated horsepower for propeller-driven aircraft or pounds thrust for turbofan aircraft. If an engine is flat rated, enabling it to produce takeoff rated output at a higher than ISA (standard day) ambient temperature, the flat rating limit is shown as ISA+XXC. Highly flat rated engines, i.e. engines that can produce takeoff rated thrust at a much higher than standard ambient temperature, typically provide substantially improved high density altitude and highaltitude cruise performance.

▶ Inspection Interval is the longest, scheduled hourly major maintenance interval for the engine, either "t" for TBO or "c" for compressor zone inspection.



OC is shown only for engines that have "on condition" repair or replace parts maintenance.

### Weights (lb.)

Weight categories are listed as appropriate to each class of aircraft.

► Max Ramp - Maximum ramp weight for taxi

► Max Takeoff - Maximum takeoff weight as determined by structural limits

Max Landing – Maximum landing weight as determined by structural limits
 Zero Fuel – Maximum zero fuel weight, shown by "c," indicating the certified MZFW or "b," a *B&CA*-computed weight based on MTOW minus the weight of fuel required to fly 1.5 hr. at high-speed cruise
 Max ramp, max takeoff and max landing weights may be the same for light aircraft that may only have a certified max takeoff weight.

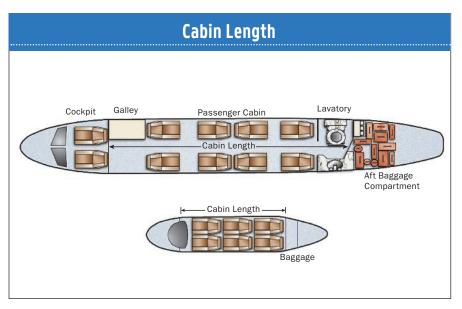
**EOW/BOW** – Empty Operating Weight is shown for piston-powered airplanes. Basic Operating Weight, in contrast, is based on the average EOW weight of the last 10 commercial deliveries, plus 200 lb. for each required crewmember. We require four crewmembers, three flight crew and one cabin attendant for ultralong-range aircraft.

Basic Operating Weight, which essentially is EOW plus required flight crew, is shown for turbine-powered airplanes. EOW is based on the factory standard weight, plus items specified in the *B&CA* Required Equipment List, less fuel and oil.

There is no requirement to add in the weight of cabin stores, but some manufacturers choose to include galley stores and passenger supplies as part of the BOW build up. Life vest, life rafts and appropriate deep-water survival equipment are included in the weight buildup of the 80,000+ lb., ultra-long-range aircraft.

▶ Max Payload – Zero Fuel weight minus EOW or BOW, as appropriate. For pistonengine airplanes, Max Payload frequently is a computed value because it is based on the *B&CA* ("b") computed maximum ZFW.

**Executive Payload** – Based on 170 lb. per occupant for multiengine piston-engine airplanes and 200 lb. per occupant for turbine-engine airplanes, as shown in the executive seating section of the "Characteristics" section. Both pilots and passengers, however, are counted as occupants in piston-engine airplanes. Only passengers are counted as occupants in turbinepowered airplanes because the required crew is included in the BOW.



If the Executive Payload exceeds the Maximum Payload, we use Maximum Payload. Executive payload is not computed for single-engine piston airplanes. Max Fuel – Usable fuel weight based on 6.0 lb. per U.S. gallon for avgas or 6.7 lb. per U.S. gallon for jet fuel. Fuel capacity includes optional, auxiliary and long-range tanks, unless otherwise noted.

► Available Payload With Full Fuel – Max Ramp weight minus the tanks-full weight, not to exceed Zero Fuel weight minus EOW or BOW.

► Available Fuel With Maximum Payload - Maximum Ramp weight minus Zero Fuel weight, not to exceed maximum fuel capacity.

► Available Fuel With Executive Payload — Available fuel weight based on max ramp minus BOW plus executive payload, up to the actual fuel capacity.

### Limits

*B&CA* lists V speeds and other limits as appropriate to the class of airplane. These are the abbreviations used on the charts:

► VNE - Never exceed speed (red line for piston-engine airplanes).

 VNO — Normal operating speed (top of the green arc for piston-engine airplanes).
 VMO — Maximum operating speed (red line for turbine-powered airplanes).

▶ MMO – Maximum operating Mach number (red line for turbofan-powered airplanes and a few turboprop airplanes).

 FL/Vmo – Transition altitude at which VMo equals MMo (large turboprop and turbofan aircraft).

► VA - Maneuvering speed (except for

certain large turboprop and all turbofan aircraft).

► VDEC — Accelerate/stop decision speed (multiengine piston and light multiengine turboprop airplanes).

▶ VMCA — Minimum control airspeed, airborne (multiengine piston and light multiengine turboprop airplanes).

► Vso - Maximum stalling speed, landing configuration (single-engine airplanes)

► Vx - Best angle-of-climb speed (singleengine airplanes).

**Vxse** – Best angle-of-climb speed, oneengine inoperative (multiengine piston and multiengine turboprop airplanes under 12,500 lb.).

► Vy - Best rate-of-climb speed (singleengine airplanes).

▶ VYSE — Best rate-of-climb speed, oneengine inoperative (multiengine piston and multiengine turboprop airplanes under 12,500 lb.).

► V2 - Takeoff safety speed (large turboprops and turbofan airplanes).

▶ VREF — Reference landing approach speed (large turboprops and turbofan airplanes, four passengers, NBAA IFR reserves; eight passengers for ultra-longrange aircraft).

**PSI** – Cabin pressure differential (all pressurized airplanes).

### **Airport Performance**

► Approved Flight Manual takeoff runway performance is shown for sealevel, standard day and for 5,000-ft. elevation/25C day density altitude. Allengine takeoff distance (TO) is shown for single-engine and multiengine piston, and turboprop airplanes with an MTOW



of less than 12,500 lb. Takeoff distances and speeds assume Maximum Takeoff weight, unless otherwise noted.

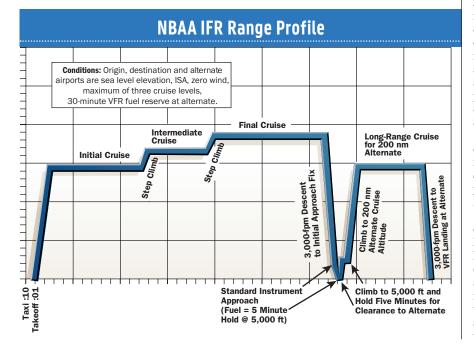
► Accelerate/Stop distance (A/S) is shown for small multiengine piston and small turboprop airplanes. Takeoff field length (TOFL), the greater of the one-engine inoperative (OEI) takeoff distance or the accelerate/stop distance, is shown for FAR Part 23 Commuter Category and FAR Part 25 airplanes. If the accelerate/stop and accelerate/stop distances are equal, the TOFL is the balanced field length.

► Landing distance (LD) is shown for FAR Part 23 Commuter Category and FAR Part 25 Transport Category airplanes. The landing weight is BOW plus four passengers and NBAA IFR fuel reserves. We assume that 80,000+ lb. ultra-long-range aircraft will have eight passengers on board.

The V2 and VREF speeds are useful for reference when comparing the TOFL and LD numbers because they provide an indication of potential minimum-length runway performance when low RCR or runway gradient is a factor.

*B&CA* lists two additional numbers for large turboprop- and turbofan-powered airplanes. First, we publish the mission weight, which is the lower of: (1) the actual takeoff weight with four passengers (eight passengers for ultra-long-range aircraft) and full fuel when departing from a 5,000-ft./25C airport or (2) the maximum allowable takeoff weight when departing with the same passenger load and at the same density altitude.

For two-engine aircraft, the mission



weight, when departing from a 5,000ft./ISA+20C airport, may be less than the MTOW because of FAR Part 25 second-segment, one-engine-inoperative, climb performance requirements. Aircraft with highly flat-rated engines are less likely to have a Mission Weight that is performance limited when departing from hot and high airports.

For three-engine aircraft, the mission weight usually is based on full tanks and the actual number of passengers, rather than being performance limited.

Second, we publish the NBAA IFR range for the hot and high departure mission weight, assuming a transition into standard day, ISA flight conditions after takeoff. For purposes of computing NBAA IFR range, the aircraft is flown at the long-range cruise speed shown in the "Cruise" block or at the same speed as shown in the "Range" block.

### Climb

The all-engine time to climb provides an indication of overall climb performance, especially if the aircraft has an all-engine service ceiling well above our sample topof-climb altitudes. We provide the all-engine time to climb to one of three specific altitudes, based on type of aircraft departing at MTOW from a sea-level, standard-day airport: (1) FL 100 (10,000 ft.) for normally aspirated single-engine and multiengine piston aircraft, plus pressurized single-engine piston aircraft and unpressurized turboprop aircraft; (2) FL 250 for pressurized single-engine and multi-engine turboprop aircraft; or (3) FL 370 for turbofan-powered aircraft. These data are published as time-to-climb in minutes/climb altitude. For example, if a non-pressurized twin-engine piston aircraft can depart from a sea-level airport at MTOW and climb to 10,000 ft. in 8 min., the time to climb is expressed as 8/FL 100.

We also publish the initial all-engine climb feet per nautical mile gradient, plus initial engine-out climb rate and gradient, for single-engine and multiengine pistons and turboprops with MTOWs of 12,500 lb. or less.

The one-engine-inoperative (OEI) climb rate for multi-engine aircraft at MTOW is derived from the Airplane Flight Manual. OEI climb rate and gradient is based on landing gear retracted and wing flaps in the takeoff configuration used to compute the published takeoff distance. The climb gradient for such airplanes is obtained by dividing the product of the climb rate (fpm) in the



Airplane Flight Manual times 60 by the Vy or Vyse climb speed, as appropriate.

The OEI climb gradients we show for FAR Part 23 Commuter Category and FAR Part 25 Transport Category aircraft are the second-segment net climb performance numbers published in the AFMs. Please note: The AFM net secondsegment climb performance numbers are adjusted downward by 0.8% to compensate for variations in pilot technique and ambient conditions.

The OEI climb gradient is computed at the same flap configuration used to calculate the takeoff field length.

### Ceilings (ft.)

▶ Maximum Certificated Altitude — Maximum allowable operating altitude determined by airworthiness authorities.

► All-Engine Service Ceiling — Maximum altitude at which at least a 100-fpm rate of climb can be attained, assuming the aircraft departed a sea-level, standard-day airport at MTOW and climbed directly to altitude.

▶ OEI (Engine Out) Service Ceiling – Maximum altitude at which a 50-fpm rate of climb can be attained, assuming the aircraft departed a sea-level, standard-day airport at MTOW and climbed directly to altitude.

**Sea-Level Cabin (SLC) Altitude** – Maximum cruise altitude at which a 14.7-psia, sea-level cabin altitude can be maintained in a pressurized airplane.

### Cruise

Cruise performance is computed using EOW with four occupants or BOW with four passengers and one-half fuel load. Ultra-long-range aircraft carry eight passengers for purposes of computing cruise performance.

Assume 170 lb. for each occupant of a piston-engine airplane and 200 lb. for each occupant of a turbine-powered aircraft.

► Long Range – True Air Speed (TAS), fuel flow in pounds/hour, flight level (FL) cruise altitude and specific range for long-range cruise by the manufacturer.

▶ **Recommended** (Piston-Engine Airplanes) — TAS, fuel flow in pounds/ hour, FL cruise altitude and specific range for normal cruise performance specified by the manufacturer.

▶ High Speed —TAS, fuel flow in pounds/ hour, FL cruise altitude and specific range for short-range, high-speed performance specified by the aircraft manufacturer.

▶ Speed, fuel flow, specific range and altitude in each category are based on one mid-weight cruise point and these data reflect standard day conditions. They are not an average for the overall mission and they are not representative of the above standard day temperatures at cruise altitudes commonly encountered in everyday operations.

*B&CA* imposes a 12,000-ft. maximum cabin altitude requirement on CAR3/ FAR Part 23 normally aspirated aircraft. Turbocharged airplanes are limited to FL 250, providing they are fitted with supplemental oxygen systems having sufficient capacity for all occupants for the duration of the mission. Pressurized CAR3/FAR Part 23 aircraft are limited to a maximum cabin altitude of 10,000 ft. For FAR Part 23 commuter Category and FAR Part 25 aircraft, the maximum cabin altitude for computing cruise performance is 8,000 ft.

To conserve space, we use flight levels (FL) for all cruise altitudes, which is appropriate considering that we assume standard day ambient temperature and pressure conditions. Cruise performance is subject to B&CA's verification.

### Range

*B&CA* shows various paper missions for each aircraft that illustrate range versus payload tradeoffs, runway and cruise performance, plus fuel efficiency. Similar to the cruise profile calculations, *B&CA* limits the maximum altitude to 12,000 ft. for normally aspirated, non-pressurized CAR3/FAR Part 23 aircraft, 25,000 ft. for turbocharged airplanes with supplemental oxygen, 10,000 ft. cabin altitude for pressurized CAR 3/FAR Part 23 airplanes and 8,000 ft. cabin altitude for FAR Part 23 Commuter Category or FAR Part 25 aircraft.

► Seats-Full Range (Single-Engine Piston Airplanes) — Based on typical executive configuration with all seats filled with 170 lb. occupants, with maximum available fuel less 45-min. IFR fuel reserves. We use the lower of seats full or maximum payload.

► Tanks Full Range (Single-Engine Piston Airplanes) — Based on one 170-lb. pilot, full fuel less 45-min. IFR fuel reserves.

► Executive Payload (Multiengine Piston Airplanes and Single-Engine Turboprops) — Based on typical executive configuration with all seats filled with 170-lb. occupants, maximum available fuel less 45-min. IFR fuel reserves. We use the lower of seats full or maximum payload.

▶ Maximum Fuel With Available Payload (Single-Engine Turboprops) —Based on BOW, plus full fuel and the maximum available payload up to maximum ramp weight. Range is based on arriving at destination with NBAA IFR fuel reserves, but only a 100-mi. alternate is required.

► Ferry (Multiengine Piston Airplanes and Single-Engine Turboprops) — Based on one

170-lb. pilot, maximum fuel less 45-min. IFR fuel reserves.

Please note: None of the missions for piston-engine aircraft includes fuel for diverting to an alternate. However, singleengine turboprops are required to have NBAA IFR fuel reserves, but only a 100 mi. alternate is required.

NBAA IFR range format cruise profiles, having a 200 mi. alternate, are used for FAR Part 25 Transport Category turbine-powered aircraft. In the case of FAR Part 23 turboprops, including those certified in the Commuter Category, and FAR Part 23 turbofan aircraft, only a 100 mi. alternate is needed. The difference in alternate requirements should be kept in mind when comparing range performance of various classes of aircraft.

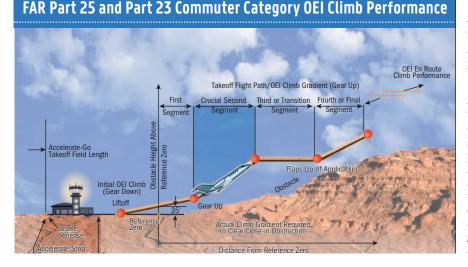
► Available Fuel With Maximum Payload (Multiengine Turbine Airplanes) —Based on aircraft loaded to maximum zero fuel weight with maximum available fuel up to maximum ramp weight, less NBAA IFR fuel reserves at destination.

► Available Payload With Maximum Fuel (Multiengine Turbine Airplanes) –Based on BOW plus full fuel and maximum available payload up to maximum ramp weight. Range based on NBAA IFR reserves at destination.

► Full/Maximum Fuel With Four Passengers (Multiengine Turbine Airplanes) —Based on BOW plus four 200-lb. passengers and the lesser of full fuel or maximum available fuel up to maximum ramp weight. Ultra-long-range aircraft must have eight passengers on board.

▶ Ferry (Multiengine Turbine Airplanes) — Based on BOW, required crew and full fuel, arriving at destination with NBAA IFR fuel reserves.

We allow 2,000-ft. increment step



### **NOTICE TO READERS**

During recent years, the U.S. Federal Trade Commission has conducted investigations into the practice of certain industries in fixing and advertising list prices. It is the position of the FTC that it is deceptive to the public and against the law for list prices of any product to be specified or advertised in a trade area if the majority of sales are made at less than those prices.

*B&CA* is not in a position to know the prices for most of the sales in each trading area in the United States for each of the products in this issue. Therefore, the prices shown in the tables and text in the *Purchase Planning Handbook* are based on suggested list prices furnished to us by the manufacturers or distributors, or on prices estimated by the editors. It may be possible to purchase some items in your trading area at prices less than those reported in this issue of *B&CA*. Also, almost all manufacturers and distributors caution that prices are subject to change without notice.

climbs above the initial cruise altitude to improve specific range performance, even though current air traffic rules in North America provide for 4,000-ft. altitude semicircular directional traffic separation above FL 290. The altitude shown in the range section is the highest cruise altitude for the trip — not the initial cruise or mid-mission altitude.

The range profiles are in nautical miles, and the average speed is computed by dividing that distance by the total flight time or weight-off-wheels time en route. The Fuel Used or Trip Fuel includes the fuel consumed for start, taxi, takeoff, cruise, descent and landing approach but not after-landing taxi or reserves.

The Specific Range is obtained by dividing the distance flown by the total fuel burn. The Altitude is the highest cruise altitude achieved on the specific mission profile shown.

### Missions

Various paper missions are computed to illustrate the runway requirements, speeds, fuel burns and specific range, plus cruise altitudes. The mission ranges are chosen to be representative for the airplane category. All fixed-distance missions are flown with four passengers on board, except for ultra-long-range airplanes, which have eight passengers on board. The pilot is counted as a passenger on board piston-engine

### **B&CA Required Equipment List**

airplanes. If an airplane cannot complete a specific fixed distance mission with the appropriate payload, *B&CA* shows a reduction of payload in the remarks section or marks the fields NP (Not Possible) at our option.

Runway performance is obtained from the Approved Airplane Flight Manual. Takeoff distance is listed for single-engine airplanes; accelerate/stop distance is listed for piston twins and light turboprops; and takeoff field length, which often corresponds to balanced field length, is used for FAR Part 23 Commuter Category and FAR Part 25 large Transport Category airplanes.

Flight Time (takeoff to touchdown, or weight-off-wheels, time) is shown for turbine airplanes. Some piston-engine manufacturers also include taxi time, resulting in a chock-to-chock, Block Time measurement. Fuel Used, though, is the actual block fuel burn for each type of aircraft, but it does not include fuel reserves. The cruise altitude shown is that which is specified by the manufacturer for fixed-distance missions.

▶ 200 nm – (Piston-engine airplanes)

500 nm – (Piston-engine airplanes)
 300 nm – (Turbine-engine airplanes,

except ultra-long range)

▶ 600 nm – (Turbine-engine airplanes, except ultra-long range)

▶ 1,000 nm – (All turbine-engine airplanes)

▶ 3,000 nm – (Ultra-long-range turbineengine airplanes)

▶ 6,000 nm — (Ultra-long-range turbineengine airplanes)

### Remarks

In this section, *B&CA* generally includes the base price, if it is available or applicable; the certification basis and year; and any notes about estimations, limitations or qualifications regarding specifications, performance or price. All prices are in 2014 dollars, FOB at a U.S. delivery point, unless otherwise noted. The certification basis includes the regulation under which the airplane was originally type certified, the year in which it was originally certified and, if applicable, subsequent years during which the airplane was re-certified.

### General

Abbreviations are used throughout the tables: "NA" means not available; "—" indicates the information is not applicable; and "NP" signifies that specific performance is not possible. **B&CA** 

Primary flight instruments

	Jets ≥20,000 lb
	Jets <20,000 lb
	Turboprops >12,500 lb
	Turboprops ≤12,500 lb
	Single-Engine Turboprops
Multien	ngine Pistons, Turbocharged
	Multiengine Pistons
Single-Engine Pisto	-
Single-Engine Pistons, Turl	
Single-Engine Pis	istons
POWERPLANT SYSTEMS	
Batt temp indicator (nicad only, for each battery) Engine synchronization	
Fire detection, each engine	
Fire extinguishing, each engine	• • •
Propeller, reversible pitch Propellers, synchronized	••••
Thrust reversers/attenuators	
AVIONICS	
ADF Air data computer	
Altitude alerter	
Altitude encoder	
Antennas, headsets, microphones Audio control panel	
Automatic flight guidance, 2-axis, alt hold	
Automatic flight guidance, 3-axis, alt hold DME	
EFIS	
ELT	
Flight director FMS, TSO C115 or GPS, TSO C129 IFR approach	
Glideslope receiver	
HSI, slaved (or equivalent EFIS function)	
Marker beacon receiver Radio altimeter	
Radiotelephone	
RMI (or equivalent function on EFIS display) RVSM certification	
TAWS	• •
TCAS I/II (FAR Part 25 airplanes only)	••
Transponder VHF comm, 25-kHz spacing	
VHF comm, 8.33-kHz spacing	
VHF nav, 360-channel Weather radar	
GENERAL	
Air conditioning, vapor cycle (not required with APU)	
Anti-skid brakes APU (required for air-start engines, ACM air conditioning)	
Cabin/cockpit dividers	• • •
Corrosion-proofing, internal	
Exterior paint, tinted windows Fire extinguisher, cabin	
Fire extinguisher, cockpit	
Fuel tanks, long-range	
Ground power jack Headrests, air vents, all seats	
Lavatory	
Lights, strobe/anti-collision beacon, navigation, landing/tax Lights, internally lighted instrument, cockpit flood, courtesy	
Oxygen, supplemental, all seats	
Refreshment center	
Seats, crew, articulating Seats, passenger, reclining	
Shoulder harness, all seats and crew with inertia reel	
Tables, cabin work ICE AND RAIN PROTECTION	
Alternate static pressure source (not required with 2 DADC)	
Approval, flight into known icing	
Ice protection plates Pitot heat	
Static wicks	
Windshield rain removal, mechanical or repellent coating	
INSTRUMENTATION Angle-of-attack stall margin indicator	• •
EGT	
IVSI (or equivalent EFIS, DADC function)	
Outside air temperature gauge Primary flight instruments	

Required

Dual required

### SINGLE-ENGINE PISTONS NORMALLY ASPIRATED

Manufacturer			Cirrus Design	Piper	Cirrus Design	Mooney	GA8 Airvan (Pty) Ltd.	Textron Aviation	
Model			SR20	Arrow PA-28R-201	SR22	Ovation3 M-20R	Airvan GA-8	Beech Bonanza G36 G36	
B&CA Equippe	ed Price		\$349,900	\$462,400	\$489,900	\$649,000	\$726,960	\$777,285	
		Seating	1+3/5	1+3/3	1+3/5	1+3/4	1+6/7	1+4/5	
Character-		Wing Loading	21.0	16.2	23.5	19.3	20.7	20.2	
istics		Power Loading	15.25	13.75	11.61	10.86	13.33	12.17	
Eutomol		Noise (dBA)	83.4 26.0	77.7 24.7	83.7 26.0	NA 26.9	84.9 29.3	76.7 27.5	
External Dimensions		Length Height	8.9	7.9	8.9	8.3	12.8	8.6	
(ft.)	Spar		38.3	35.4	38.3	36.1	40.7	33.5	
Internal		Length	8.0	7.7	8.0	8.1	11.6	12.6	
Dimensions		Height	4.1	3.7	4.1	3.7	3.7	4.2	
(ft.)	Height Width		4.1	3.5	4.1	3.6	4.2	3.5	
	Width Engine		Cont IO-360-ES	Lyc 10-360-C1C6	Cont IO-550-N	Cont IO-550-G-AP	Lyc 10-540-K1A5	Cont IO-550-B	
Power		Output (hp)	200	200	310	310	300	300	
		Inspection Interval	2,000t	2,000t	2,000t	2,000t	2,000t	1,900t	
		Max Ramp	3,050	2,758	3,600	3,374	4,014	3,663	
		Max Takeoff	3,050	2,750	3,600	3,368	4,000	3,650	
	Max Landing		3,050	2,750	3,600	3,200	4,000	3,650	
		Zero Fuel	2,900c	2,636b	3,400c	3,197b	3,849b	3,494b	
\A/-:		EOW	2,128	1,798	2,260	2,260	2,241	2,630	
Weights (lb.)		Max Payload	772	838	1,140	937	1,608	864	
		Useful Load Max Barrare	922 130	960 200	1,340 130	1,114 120	1,773 180	1,033 670	
		Max Baggage	336	432	552	600	540	444	
	Max Fuel Available Payload w/Max Fuel		586	528	788	514	1,233	589	
		Available Fuel w/Max Payload	150	122	200	177	165	169	
		VNE	204	183	205	195	185	203	
Limits	VNO		166	146	176	174	143	165	
		VA	131	118	140	127	121	139	
		TO (SL elev./ISA temp.)	2,221	1,600	1,756	1,620	1,860	1,913	
Airport		TO (5,000' elev.@25C)	3,752	3,250	3,016	2,500	3,670	3,450	
Perfor-		Vso .	61	55	64	59	57	59	
mance		Vx .	83 96	78	88	80 105	70	84	
		Vy Time to Climb (min.)/Altitude	96 16/FL 100	90 16/FL 100	108 11/FL 100	105 10/FL 100	86 15/FL 100	100 14/FL 100	
Climb		Initial Gradient (ft./nm)	581	560	775	NA	787	730	
Ceiling (ft.)		Service	17,500	16,200	17,500	NA	20,000	18,500	
		TAS	140	124	160	163	127	160	
	Long	Fuel Flow	50	51	68	50	78	71	
	Range	Altitude	FL 080	FL 100	FL 080	FL 120	FL 120	FL 080	
		Specific Range	2.800	2.431	2.353	3.260	1.628	2.254	
		TAS	152	130	171	186	135	167	
Cruise	Recommended	Recommended	Fuel Flow	63	68	92	84	88	86
		Altitude	FL 080	FL 090	FL 080	FL 120	FL 080	FL 080	
		Specific Range TAS	2.413 159	1.912 137	1.859 180	2.214 196	1.534 142	1.942 174	
		Fuel Flow	70	76	107	198	142	94	
	High Speed	Altitude	FL 80	FL 060	FL 80	FL 120	FL 060	FL 080	
		Specific Range	2.271	1.803	1.682	1.719	1.406	1.851	
		Nautical Miles	584	537	1,118	1,075	487	242	
		Average Speed	140	121	162	161	124	153	
		Fuel Used	210	156	492	438	339	126	
Ranges		Specific Range/Altitude	2.781/FL 080	3.442/FL 070	2.272/FL 080	2.454/FL 120	1.437/FL 120	1.921/FL 040	
		Nautical Miles	764	926	1,118	1,465	690	919	
		Average Speed	140	121	162	173	125	151	
		Fuel Used	275	408	492	558	464	403	
		Specific Range/Altitude	2.778/FL 080	2.270/FL 070	2.272/FL 080	2.625/FL 120	1.487/FL 120	2.280/FL 120	
		Runway Block Time	1,446 1+18	1,600 1+29	1,303	1,230 1+13	1,860 1+38	1,694 1+10	
	200 nm	Fuel Used	100	125	127	115	157	132	
Missions		Specific Range/Altitude	2.000/FL 080	1.600/FL 070	1.575/FL 080	1.739/FL 050	1.274/FL 120	1.515/FL 040	
(4 occupants)		Runway	1,446	1,600	1,519	1,290	1,860	2,054	
	500 mm	Block Time	3+17	3+50	2+49	2+58	3+55	2+54	
	500 nm	Fuel Used	226	278	305	221	339	304	
		Specific Range/Altitude	2.212/FL 080	1.799/FL 090	1.639/FL 080	2.262/FL 100	1.475/FL 120	1.645/FL 060	
		Suggested Base Price	\$349,900	\$448,750	\$489,900	\$649,000	\$726,960	\$770,385	
Remarks		Certification Basis	FAR 23, 1999 Includes Garmin Perspective avionics.	CAR 3, 1976/2001 Garmin G500 standard.	FAR 23, 2000 Includes Garmin Perspective avionics; MTOW increased by 200 lb.	CAR 3/FAR 23, 1955/94; STC SA02483CH Includes Garmin G1000 with GFC700 autopilot standard.	FAR 23 A 54 Includes Garmin G500 All data preliminary.	CAR 3, 1956/69/83/2005 Restyled interior; new, lightweight A/C system; Garmin G1000 with SVS.	

### SINGLE-ENGINE PISTONS TURBOCHARGED

Manufacturer		Textron Aviation Cessna Skylane JT-A	Cirrus SR22T	Textron Aviation Cessna Turbo Stationair	Mooney Acclaim Type S	GA8 Airvan (Pty) Ltd. GA8 Airvan TC	Textron Aviation Cessna Corvalis TT <sup>x</sup>	
			CE-J182T	SR 22	CE-T206H	M0-20TN	GA8-320 TC	LC41-550-FG
B&CA Equippe	ed Price	Contine	\$530,000	\$589,900	\$615,000	\$699,000	\$761,030	\$799,000
Character-		Seating Wing Loading	<u> </u>	1+3/5 23.5	1+5/5 20.7	1+3/3 19.2	1+6/7 20.7	1+3/3 25.5
istics		Power Loading	13.66	11.43	11.61	12.03	13.13	11.61
131103		Noise (dBA)	NA	80.3	75.8	78.0	85.4	81.5
External		Length	28.5	26.0	28.3	26.9	28.3	25.2
Dimensions		Height	9.3	8.9	9.3	8.3	9.3	9.0
(ft.)		Span	36.0	38.3	36.0	36.4	36.0	36.0
Internal		Length	7.2	8.0	9.3	8.1	11.6	7.9
Dimensions		Height	4.0	4.1	4.1	3.7	3.7	4.1
(ft.)		Width	3.5	4.1	3.7	3.6	4.2	4.0
		Fastar	SMA	Cont	Lyc	Cont	Lyc	Cont
Power		Engine	SR305-230E-C1	TSI0-550-K	TIO-540-AJ1A	TSI0-550-G	TIO-540-AH1A	TSI0-550-C
i owei		Output (hp)	227	315	310	280	320	310
		Inspection Interval	NA	2,000t	2,000t	2,000t	1,800t	2,000t
		Max Ramp	3,110	3,609	3,617	3,374	4,214	3,600
		Max Takeoff	3,100	3,600	3,600	3,368	4,200	3,600
		Max Landing	2,950	3,600	3,600	3,200	4,000	3,420
		Zero Fuel	NA	3,400c	3,429b	3,173b	4,053b	3,300c
M/oighto (III)		EOW May Davided d	2,092	2,342	2,336	2,378	2,349	2,600
Weights (lb.)		Max Payload	NA	1,058	1,093	795	1,704	700
		Useful Load	1,018	1,267	1,281	996	1,865	1,000
		Max Baggage	200	130	180	120	180	120
		Max Fuel	583	552	522	612	540	612
		ailable Payload w/Max Fuel	435	715	759	384	1,325	388
	AVa	ailable Fuel w/Max Payload	NA	209	188	201	161	300
Linsite		VNE	NA	205	182 149	195 174	185 143	230 181
Limits		Vno Va	NA	176 140	149	174	143	181
		TO (SL elev./ISA Temp.)	NA 1,385	140	1,740	2,100	1,840	1,900
Airport		TO (5,000' elev.@25C)		2,268	2,470	3,100	2,788	2,460
Airport Perfor-		Vs0	NA	64	57	60	61	61
mance		Vsc Vx	NA	88	69	80	71	82
manoo		Vy	NA	103	89	105	81	110
	Tir	me to Climb (min.)/Altitude	NA/FL 100	7/FL 100	11/FL 100	7/FL 100	13/FL 100	7/FL 100
Climb		Initial Gradient (ft./nm)	NA	782	724	770	825	701
		Certificated	20,000	25,000	25,000	25,000	20,000	25,000
Ceilings (ft.)		Service	20,000	25,000	27,000	25,000	20,000	25,000
		TAS	NA	171	137	215	125	208
	Long	Fuel Flow	NA	76	85	99	68	78
	Range	Altitude	FL 200	FL 250	FL 240	FL 250	FL 200	FL 250
		Specific Range	NA	2.250	1.612	2.172	1.838	2.667
		TAS	NA	201	155	227	130	227
Cruise		Fuel Flow	NA	98	99	128	78	130
Cluise		Altitude	FL 200	FL 250	FL 240	FL 180	FL 200	FL 250
		Specific Range	NA	2.051	1.566	1.773	1.667	1.746
		TAS	156	213	164	242	135	235
	High Speed	Fuel Flow	NA	110	114	130	98	152
		Altitude	FL 100	FL 250	FL 200	FL 250	FL 200	FL 250
		Specific Range	NA	1.936	1.439	1.862	1.378	1.546
		Nautical Miles	NA	1,021	200	500	233	450
		Average Speed	NA	171	139	178	125	199
		Fuel Used	NA	486	185	259	220	262
Ranges		Specific Range/Altitude	NA/NA	2.101/FL 250	1.081/FL 200	1.931/FL 160	1.059/FL 200	1.718/FL 250
		Nautical Miles	1,025	1,021	618	1,122	618	1,189
		Average Speed	NA	171	137	200	125	204
		Fuel Used	NA NA (EL 140	486	459	539	459	539
		Specific Range/Altitude	NA/FL 140	2.101/FL 250	1.346/FL 240	2.082/FL 250	1.346/FL 200	2.206/FL 250
		Runway	NA	1,405	1,743	1,300	1,743	1,900
	200 nm	Block Time	NA	1+08	1+25	1+05	1+35	1+05
Miccierc		Fuel Used	NA	197	170	139	125	150
Missions		Specific Range/Altitude	NA/NA	1.015/FL 100	1.176/FL 120	1.439/FL 120	1.600/FL 120	1.333/FL 120
(4 occupants)		Runway Block Time	NA	1,699	1,743	1,380	1,743	1,900
	500 nm	Block Time	NA	2+28	3+21	2+54	3+30	2+24
		Fuel Used	NA	360	373	259	373	336
		Specific Range/Altitude	NA/NA	1.389/FL 180	1.340/FL 240	1.931/FL 250	1.340/FL 200	1.488/FL 250
Remarks		Suggested Base Price Certification Basis	\$530,000 FAR 23 pending All data preliminary; Includes Garmin G1000 with GFC700.	\$544,900 FAR 23, 2010 Includes Perspective Global Connect worldwide wx; 200-lb. MTOW increase.	\$615,000 FAR 23, 1998 Includes Garmin G1000 with GFC700 autopilot.	\$699,000 CAR 3, 1955/89/06 Includes Garmin G1000 with GFC700 autopilot standard.	\$597,500 FAR 23, 1998 Includes Garmin G1000 with GFC700 autopilot; All data preliminary.	\$799,000 FAR 23 Includes Garmin G2000 with GFC700 autopilot.

### SINGLE-ENGINE PISTONS PRESSURIZED

Manufacturer			Piper Aircraft	Piper Aircraft
Model			Malibu Matrix PA-46R-350	Malibu Mirage PA-46-350P
B&CA Equippe	d Price		\$1,014,950	\$1,178,875
		Seating	1+4/5	1+4/5
Character-		Wing Loading	24.8	24.8
istics		Power Loading	12.40	12.40
Extornal		Noise (dBA) Length	81.0 28.9	81.0 28.9
External Dimensions		Height	11.3	11.3
(ft.)		Span	43.0	43.0
Internal		Length	12.4	12.4
Dimensions		Height	3.9	3.9
(ft.)		Width	4.2	4.2
		Engine	Lyc TIO-540-AE2A	Lyc TIO-540-AE2A
Power		Output (hp)	350	350
		Inspection Interval	2,000t	2,000t
		Max Ramp	4,358	4,358
		Max Takeoff	4,340	4,340
		Max Landing	4,123	4,123
		Zero Fuel	4,123c	4,123c
Moighte (lb)		EOW .	2,969	3,146
Weights (lb.)		Max Payload Useful Load	1,154	977 1,212
		Max Baggage	200	200
		Max Daggage Max Fuel	720	720
	ļ	vailable Payload w/Max Fuel	669	492
	4	vailable Fuel w/Max Payload	235	235
		VNE	198	198
Limits		VNO	168	168
		VA	133	133
		PSI TO (SL elev./ISA temp.)	5.5	5.5 2,090
		TO (5,000' elev.@25C)	2,977	2,030
Airport		Vso	58	58
Performance		Vx	81	81
		Vy	110	110
Climb		Time to Climb (min.)/Altitude	8/FL 100	8/FL 100
		Initial Gradient (ft./nm)	703	703
Collings (ft.)		Certificated Service	25,000	25,000 25,000
Ceilings (ft.)		Sea-Level Cabin		12,300
		TAS	156	156
	Long	Fuel Flow	66	66
	Range	Altitude	FL 250	FL 250
		Specific Range	2.364	2.364
		TAS	203	203
Cruise		Fuel Flow	108	108 FL 250
		Altitude Specific Range	FL 250 1.880	1.880
		TAS	213	213
	111-4-0	Fuel Flow	120	120
	High Speed	Altitude	FL 250	FL 250
		Specific Range	1.775	1.775
		Nautical Miles	867	535
		Average Speed	151	138
Ranges		Fuel Used Specific Range/Altitude	457 1.897/FL 200	312 1.715/FL 120
Hanges		Nautical Miles	1,343	1,343
		Average Speed	158	159
		Fuel Used	658	670
		Specific Range/Altitude	2.041/FL 250	2.004/FL 250
		Runway	2,090	2,090
	200 nm	Block Time	1+07	1+06
Minoiome		Fuel Used	168	167
Missions (4 occupants)		Specific Range/Altitude Runway	1.190/FL 120 2,090	1.198/FL 200 2,090
(+ occupants)		Block Time	2,090	2,090
	500 nm	Fuel Used	350	350
		Specific Range/Altitude	1.429/FL 250	1.429/FL 250
		Suggested Base Price	\$939,950	\$1.100,450
Remarks		Certification Basis	FAR 23, 1983/88 Garmin G1000 standard; FIKI optional.	FAR 23, 1983/88 Garmin G1000 with SVS and FIKI standard.
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### **MULTIENGINE PISTONS** NORMALLY ASPIRATED

Manufacturer			Vulcanair SpA	Vulcanair SpA
Model			P.68C	Victor
	d Drico		P 68C	P 68R
B&CA Equippe	a Price	Section	\$960,000 1+5/6	\$978,000 1+5/6
Character-		Seating Wing Loading		21.6
istics		Power Loading		10.80
Sucs		Noise (dBA)	74.7	74.7
External		Length	31.3	31.3
Dimensions		Height		11.2
ft.)		Span	39.4	39.4
nternal		Length	10.6	10.6
Dimensions		Height	3.9	3.9
ft.)		Width	3.8	3.8
(10.)			2 Lyc	2 Lyc
_		Engines	IO-360-A1B6	10-360-A1B6
Power		Output (hp each)	200	200
		Inspection Interval	2,000t	2,000t
		Max Ramp	4,630	4,357
		Max Takeoff	4,594	4,321
		Max Landing	4,365	4,321
		Zero Fuel	4,167c	4,147b
		EOW	3,153	3,197
Neights (lb.)		Max Payload	1,014	950
weights (ib.)		Useful Load	1,477	1,160
		Executive Payload	1,020	1,020
		Max Fuel	1,063	1,063
		Payload - Max Fuel	415	98
		Available Fuel w/Max Payload	463	210
	Avail	able Fuel w/Executive Payload	457	140
		Vne	194	193
imits		VNO	154	153
		VA	132	125
		TO (SL elev./ISA Temp.)	1,312	1,260
		TO (5,000' elev.@25C)	4,000	4,000
		A/S (SL elev./ISA)	2,149	2,300
Airport		A/S (5,000' elev.@25C)	2,854	4,025
Performance		Vmca	60	60
onioninanoo		VDEC	70	70
		Vxse	82	82
		Vyse	88	88
		Time to Climb (min.)/Altitude	12/FL 100	12/FL 100
		Initial Engine-Out Rate (fpm)	217	217
Climb	Init	al All-Engine Gradient (ft./nm)	1,100	920
		I Engine-Out Gradient (ft./nm)	147	147
		Certificated	_	_
Ceilings (ft.)		All-Engine Service	18,000	20,000
		Engine-Out Service	5,000	5,650
		TAS	144	144
	Long	Fuel Flow	94	94
	Range	Altitude	FL 080	FL 080
		Specific Range	1.532	1.532
		TAS	155	155
Omice		Fuel Flow	108	108
Cruise		Altitude	FL 080	FL 080
		Specific Range	1.435	1.435
		TAS	162	162
	High Speed	Fuel Flow	116	116
	High Speed	Altitude	FL 080	FL 080
		Specific Range	1.397	1.397
		Nautical Miles	575	575
	Executive Deel	Average Speed	145	145
		ad Trip Fuel	395	395
Ranges		Specific Range/Altitude	1.456/FL 080	1.456/FL 080
		Nautical Miles	855	855
		Average Speed	145	145
		Trip Fuel	561	561
		Specific Range/Altitude	1.524/FL 080	1.524/FL 080
		Runway	1,450	1,450
		Block Time	1+28	1+28
lingian		Fuel Used	140	140
Missions		Specific Range/Altitude	1.429/FL 080	1.429/FL 080
4 occu-		Runway		1,500
pants)		Block Time	3+25	3+25
		Fuel Used	375	375
		Specific Range/Altitude		1.333/FL 080
		Suggested Base Price	\$960,000	\$978,000
			, , , , , , , , , , , , , , , , , , , ,	, 5. 6,000
			EAD 22 4076/00	EASA 22 2000
Remarks		Certification Basis	FAR 23, 1976/80 Garmin G950;	EASA 23, 2009 Garmin G950;
			STEC 55X DFCS.	STEC 55X DFCS.

### MULTIENGINE PISTONS NORMALLY ASPIRATED

<u>/lanufacturer</u> Model			Textron Aviation Beech Baron G58	Manufactur Model
	d Dring		G58	
&CA Equippe	a Price	Seating	\$1,394,400 1+4/5	<b>B&amp;CA</b> Equip
haracter-		Wing Loading	27.6	Character-
stics		Power Loading	9.17	istics
5105		Noise (dBA)	77.6	isucs
xternal		Length	29.8	External
imensions		Height	9.8	External
		Span		Dimensions
t.)			37.8	(ft.)
nternal		Length	12.6	Internal
imensions		Height	4.2	Dimensions
t.)		Width	3.5	(ft.)
		Engines	2 Cont 10-550-C	
ower		-		Power
		Output (hp each)	300	
		Inspection Interval	1,900t	
		Max Ramp	5,524	
		Max Takeoff	5,500	
		Max Landing	5,400	
		Zero Fuel	5,240b	
		EOW	4,030	
/eights (lb.)		Max Payload	1,210	Weights (lb.
8-100 (101)		Useful Load	1,494	
		Executive Payload	850	
		Max Fuel	1,164	
		Payload – Max Fuel	330	
		Available Fuel w/Max Payload	284	
	Av	vailable Fuel w/Executive Payload	644	
		VNE	223	
mits		VNO	195	Limits
		VA	165	LITIILS
		TO (SL elev./ISA Temp.)	2,345	
		TO (5,000' elev.@25C)	4,144	
		A/S (SL elev./ISA)	3,009	
irport		A/S (5,000' elev.@25C)	4,335	Airport
erformance		VMCA	84	
		VDEC	85	Perfor-
		Vxse	100	mance
		Vyse	101	
		Time to Climb (min.)/Altitude	10/FL 100	
		Initial Engine-Out Rate (fpm)	390	
limb	1	nitial All-Engine Gradient (ft./nm)	988	Climb
	In	itial Engine-Out Gradient (ft./nm)	232	Ciintio
		Certificated	_	
eilings (ft.)		All-Engine Service	20,688	
		Engine-Out Service	7,284	Ceilings (ft.
		TAS	185	00111150 (11.
	Long	Fuel Flow	144	
	Range	Altitude	FL 080	
		Specific Range	1.285	
		TAS	192	
		Fuel Flow	174	
ruise	Recommended	Altitude	FL 080	
		Specific Range	1.103	Cruise
		TAS	200	Gruise
		Fuel Flow	190	
	High Speed	Altitude	FL 080	
		Specific Range	1.053	
		Nautical Miles	723	
		Avorado Spood	180	
	Executive Payloa	d Trip Fuel	561	
anges		Specific Range/Altitude	1.289/FL 120	
anges		Nautical Miles	1,480	
		Average Speed	180	Ranges
	Ferry	Average Speed Trip Fuel	1,081	
		Specific Range/Altitude		
		Specific Range/Altitude Runway	1.369/FL 120	
		,	2,876	
	200 nm	Block Time		
		Fuel Used	219	
ission <u>s</u>		Specific Range/Altitude	0.913/FL 080	
		Runway	2,945	Missions
occu-			2+35	(4 occupants)
occu-	500 nm	Block Time		(+ occupants
occu-	500 nm	Fuel Used	487	
occu-	500 nm	Fuel Used Specific Range/Altitude	1.027/FL 100	
occu-	500 nm	Fuel Used		
l occu-	500 nm	Fuel Used Specific Range/Altitude	1.027/FL 100 \$1,387,500 CAR 3,	
ł occu- ants)	500 nm	Fuel Used Specific Range/Altitude	1.027/FL 100 \$1,387,500 CAR 3, 1957/69/83/2005	
lissions 4 occu- ants) emarks	500 nm	Fuel Used Specific Range/Altitude	1.027/FL 100 \$1,387,500 CAR 3,	Remarks

anufacturei	ſ		Vulcanair SpA	Piper Aircraft
lodel			P 68C-TC	Seneca V
CA Equippe	od Prico		\$1,015,000	<b>PA-34-220T</b> \$1,047,400
CA Equippe		Contine		
naracter-		Seating Wing Loading	<u>1+5/5</u> 20.7	<u>1+4/5</u> 22.8
		Power Loading	10.45	10.80
ics		Noise (dBA)	74.7	75.6
ternal		Length	37.6	28.6
mensions		Height	11.2	9.9
		Span	39.4	38.9
.) Iorraal			10.6	10.4
ternal		Length Height	3.9	3.6
mensions		Width	3.8	4.1
•)		WILLI	2 Lyc	2 Cont
		Engines	TIO-360-C1A6D	TSIO-360-RB
ower		Output (hp each)	210	220
		Inspection Interval	2,000t	1,800t
		Max Ramp	4,442	4,773
		Max Takeoff	4,387	4,750
		Max Landing	4,365	4,513
		Zero Fuel	4,140b	4,479c
		EOW	2,976	3,491
		Max Payload	1,164	988
eights (lb.)		Useful Load	1,466	1,331
		Executive Payload	1,020	850
		Max Fuel	1,146	732
		Payload - Max Fuel	320	599
		Available Fuel w/Max Payload	302	343
	Ava	ilable Fuel w/Executive Payload	446	481
		VNE	193	204
			153	164
nits		VA	126	139
		PSI		
		TO (SL elev./ISA temp.)	1,260	1,707
		TO (5,000' elev.@25C)	2,200	2,435
		A/S (SL elev./ISA)	2,300	2,510
rport		A/S (5,000' elev.@25C)	3,000	3,117
erfor-		VMCA	63	66
ance		VDEC	NA	73
		VXSE	NA	83
		Vyse	NA	88
		Time to Climb (min.)/Altitude	10/FL 100	7/FL 100
		Initial Engine-Out Rate (fpm)	240	253
imb	Ini	tial All-Engine Gradient (ft./nm)	1,400	996
		al Engine-Out Gradient (ft./nm)	NA	173
		Certificated	20,000	25,000
		All-Engine Service	20,000	25,000
eilings (ft.)		Engine-Out Service	11,500	16,500
		Sea-Level Cabin		
		TAS	144	167
	Long	Fuel Flow	100	108
	Range	Altitude	FL 080	FL 230
		Specific Range	1.440	1.546
		TAS	155	196
		Fuel Flow	125	144
ruise		Altitude	FL 080	FL 250
		Specific Range	1.240	1.361
		TAS	162	200
		Fuel Flow	150	156
	High Speed	Altitude	FL 080	FL 230
		Specific Range	1.080	1.282
		Nautical Miles	1,050	449
		Average Sneed	145	153
		Trip Fuel	942	348
inges		Specific Range/Altitude	1.115/FL 080	1.290/FL 130
		Nautical Miles	1,100	866
		Average Speed	145	160
		Trip Fuel	950	648
		Specific Range/Altitude	1.158/FL 080	1.336/FL 180
		Runway	NA	1,520
		nunwdy	IN/M	1,020

		Specific Range/Altitude	1.158/FL 080	1.336/FL 180
		Runway	NA	1,520
	200 nm	Block Time	1+28	1+10
		Fuel Used	260	213
sions		Specific Range/Altitude	0.769/FL 080	0.939/FL 120
ccupants)		Runway	NA	1,610
	500 nm	Block Time	3+25	2+41
		Fuel Used	485	476
		Specific Range/Altitude	1.031/FL 080	1.050/FL 200
		Suggested Base Price	\$1,015,000	\$1,011,300
narks		Certification Basis	FAR 23, 1982 Data <i>B&amp;CA</i> estimate; Garmin G950 glass cock- pit; STEC 55X DFGS.	FAR 23, 1971/80/97 Garmin G1000 with GFC 700 autopilot standard.

# MULTIENGINE PISTONS TURBOCHARGED Manufacturer Vulcanair SpA

Piper Aircraft

### SINGLE-ENGINE TURBOPROPS

Manufacturer Model			Textron Aviation Cessna Caravan CE-208	Quest Aircraft Kodiak Kodiak 100	Piper Aircraft Meridian PA-46-500TP	Textron Aviation Grand Caravan EX CE-208B	Socata TBM 900 TBM 700 N	Pilatus PC-12 NG PC-12/47E
<b>B&amp;CA</b> Equippe	ed Price		\$2,153,715	\$2,170,700	\$2,276,325	\$2,470,415	\$3,711,478	\$4,641,047
Eddin Eddinppe		Seating	1+9/13*	1+5/9	1+4/5	1+9/13*	1+5/6	1+7/10
Character-		Wing Loading	28.6	30.2	27.8	31.5	38.2	37.6
istics		Power Loading	11.85	9.67	10.18	10.16	8.70	8.71
		Noise (dBA)	79.0	84.4	76.8	84.1	76.4	79.3
External		Length	37.6	33.8	29.6	41.6	35.2	47.3
Dimensions		Height	14.9	15.3	11.3	14.8	14.3	14.0
(ft.)	Span		52.1	45.0	43.0	52.1	42.1	53.3
Internal		Length	12.7	15.8	12.3	16.7	15.0	16.9
Dimensions		Height	4.5	4.8	3.9	4.5	4.1	4.8
(ft.)		Width	5.3	4.5	4.1	5.3	4.0	5.0
(,			P&WC	P&WC	P&WC	P&WC	P&WC	P&WC
_		Engine	PT6A-114A	PT6A-34	PT6A-42A	PT6A-140	PT6A-66D	PT6A-67P
Power		Output (shp)/Flat Rating	675/ISA+31C	750/ISA+7C	500/ISA+55C	867/ISA+24C	850/ISA+49C	1,200/ISA+35C
		Inspection Interval	3,600t	4,000t	3,600t	3,600t	3,500t	3,500t
		Max Ramp	8,035	7,305	5,134	8,842	7,430	10,495
		Max Takeoff	8,000	7,255	5,092	8,807	7,394	10,450
		Max Landing	7,800	7,255	4,850	8,500	7,024	9,921
		Zero Fuel	7,432b	6,490c	4,850c	8,150b	6,032c	9,039c
		BOW	4,925	4,428	3,644	5,305	4,829	6,782
Noidete (II-)		Max Payload	2,507	2,062	1,206	2,845	1,203	2,257
Neights (lb.)		Useful Load	3,110	2,877	1,490	3,537	2,601	3,713
		Executive Payload	1,800	1,000	800	1,800	850	1,400
		Max Fuel	2,224	2,144	1,140	2,246	2,017	2,704
		Available Payload w/Max Fuel	886	733	350	1,291	653	1,009
		Available Fuel w/Max Payload	604	815	284	692	1,398	1,456
	Ava	ilable Fuel w/Executive Payload	1,310	1,877	690	1,737	1,751	2,313
		Умо	175	180	188	175	266	240
imits		Va .	150	143	127	148	160	163
		PSI	_	_	5.5	_	6.2	5.8
		TO (SL elev./ISA temp.)	2,055	1,468	2,438	2,160	2,380	2,650
Airport		TO (5,000' elev.@25C)	2,973	2,396	3,691	3,661	3,475	4,450
Perfor-		Vs0	61	60	69	61	65	67
		Vs.	90	73	95	86	100	120
nance		VA . Vy	107	101	125	108	124	130
		Time to Climb (min.)/Altitude	9/FL 100	9/FL 100	19/FL 250	9/FL 100	13/FL 250	21/FL 250
Climb		Initial Gradient (ft./nm)	771	915	753	816	1,000	833
		Certificated	25,000	25,000	30,000	25,000	31,000	30,000
Ceilings (ft.)		Service	25,000	25,000	30,000	25,000	31,000	30,000
Jennigs (i.t.)		Sea-Level Cabin			12,300	23,000	14,390	13,100
		TAS	157	139	179	163	252	203
	Land	Fuel Flow	281	229	135	324	252	203
	Long Range	Altitude	FL 100	FL 100	FL 280	FL 100	FL 310	FL 300
	nungo	Specific Range	0.559	0.607	1.326	0.503	0.956	0.829
Cruise		TAS	186	171	257	195	330	280
		Fuel Flow	379	326	237	438	412	491
	High Speed	Altitude	FL 100	FL 100	FL 280	FL 100	FL 290	FL 200
	Speed	Specific Range	0.491	0.525	1.066	0.445	0.801	0.570
		Nautical Miles	455	1,079	699	599	1,150	1,458
	Executive Pavlo		155	139	180	161	325	257
			876	1,907	649			2,073
			0.519/FL 100	0.566/FL 100	1.077/FL 280	1,255 0.477/FL 100	1,370 0.839/FL 310	0.703/FL 300
NBAA IFR		Specific Range/Altitude						
		Nautical Miles	965	1,079	953	857	1,514	1,606
Ranges		Average Speed	156	139	180 841	162	252	258
100-nm			1,795	1,907		1,768	1,599	2,266
lternate)		Specific Range/Altitude	0.538/FL 100	0.566/FL 100	1.133/FL 280	0.485/FL 100	0.947/FL 310	0.709/FL 300
		Nautical Miles	970	1,181	1,072	865	1,594	1,644
		Average Speed	156	142	220	162	252	264
		Trip Fuel	1,800	1,907	978	1,777	1,598	2,277
		Specific Range/Altitude	0.539/FL 100	0.619/FL 160	1.096/FL 280	0.487/FL 100	0.997/FL 310	0.722/FL 300
		Runway	1,468	937	2,250	1,345	1,765	1,650
		Flight Time	1+40	1+49	1+22	1+37	1+00	1+10
		Fuel Used	648	583	379	713	440	543
		Specific Range/Altitude	0.463/FL 100	0.515/FL 100	0.792/FL 280	0.421/FL 100	0.682/FL 280	0.552/FL 260
lissions		Runway	1,675	975	2,400	1,659	2,005	1,850
4 passen-	600 nm	Flight Time	3+17	3+38	2+32	3+09	1+55	2+18
ers)		Fuel Used	1,260	1,166	661	1,392	830	969
5013/		Specific Range/Altitude	0.476/FL 100	0.515/FL 100	0.908/FL 280	0.431/FL 100	0.723/FL 280	0.619/FL 270
		Runway	NP	1,467	2,438	NP	2,380	2,100
		Flight Time	NP	5+54	4+34	NP	3+10	3+48
	2,0001111	Fuel Used	NP	1,907	920	NP	1,320	1,509
		Specific Range/Altitude	NP/NP	0.524/FL 100	1.087/FL 280	NP/NP	0.758/FL 290	0.663/FL 280
		Suggested Base Price	\$1,899,000	\$1,975,000	\$2,219,850	\$2,199,000	\$3,512,088	\$3,850,000
Remarks		Certification Basis	FAR 23, 1984/98 *Export only; Garmin G1000 with GFC700 autopilot.	FAR 23, 2007 Normal category; includes Garmin G1000.	FAR 23 A 52 *1,000-nm, 3-passenger mission; Garmin G1000 with SVS.	FAR 23, 1986/2012 *Export only; Garmin G1000 with GFC700 autopilot.	FAR 23, 1990/2003/ 06/07/14 Pilot door std.; 5-blade prop; multi-seat; G1000; RVSM; SVT; sat phone/WX; 5-yr. system warranty.	FAR 23, 1996/2005/08 Honeywell Primus APEX; SmartView; BMW Designworks interior.

### MULTIENGINE TURBOPROPS 12,500-LB. MTOW OR LESS

lanufacturer Nodel			Evektor Outback EV-55	Vulcanair SpA Viator AP68TP-600	Textron Aviation King Air C90GTx C90GTi	GECI Aviation Caravan II F406	Textron Aviatio King Air 250 B200GT
&CA Equipped Price			\$2,250,000	\$2,772,000	\$3,891,200	\$4,250,000	\$6,105,425
		Seating	1+9/14	1+7/10	1+7/8	1+8/13	1+8/10
haracteristics		Wing Loading	37.4	31.4	34.4	38.9	40.3
naracteristics		Power Loading	9.46	9.58	9.53	9.85	7.35
		Noise (dBA)	NA	71.7	76.0	69.9	81.2
xternal		Length	46.6	37.0	35.5	39.1	43.8
imensions (ft.)		Height	<u>16.8</u> 53.2	<u>11.9</u> 39.4	14.3 53.7	13.2 49.5	14.8 57.9
		Span Length: OA/Net	20/16.5	17.2/11.9	12.4/12.4	14.2/12.8	16.7/16.7
Iternal		Height	4.5	4.1	4.8	4.3	4.8
imensions (ft.)		Width: Max/Floor	5.3/4.7	3.7/3.7	4.5/4.1	4.7/4.7	4.5/4.1
		· · · · · · · · · · · · · · · · · · ·	2 P&WC	2 RR	2 P&WC	2 P&WC	2 P&WC
		Engines	PT6A-21	250 B17C	PT6A-135A	PT6A-112	PT6A-52
ower		Output (shp each)/Flat Rating	536/ISA+15C	328/ISA+25C	550/ISA+30C	500/ISA+22C	850/ISA+370
		Inspection Interval	3,600t	3,500t	3,600t	3,600t	3,600t
		Max Ramp	10,207	6,338	10,545	9,925	12,590
		Max Takeoff	10,141	6,283	10,485	9,850	12,500
		Max Landing	10,141	6,283	9,832	9,360	12,500
		Zero Fuel	9,810c	5,621c	9,378c	8,500c	11,000c
		BOW	5,965	3,770	7,250	5,732	8,790
eights (lb.)		Max Payload	3,845	1,851	2,128	2,768	2,210
		Useful Load	4,242	2,568	3,295	4,193	3,800
		Executive Payload	1,800	1,400	1,400	1,600	1,600
		- Max Fuel Available Payload w/Max Fuel	3,413 829	1,487 1,081	2,573 722	3,183	3,645 155
		Available Payload w/ Max Fuel Available Fuel w/ Max Payload	397	1,081 717	1,167	1,010	1,590
		Ible Fuel w/Executive Payload	2,442	1,168	1,107	2,593	2,200
	AVdild	VMO	2,442	200	226	2,595	2,200
nits		VNO - VA	140	157	169	160	182
		PSI			5.0		6.5
		TO (SL elev./ISA temp.)	1,378	2,034	2,552	2,964	2,111
		TO (5,000' elev.@25C)	1,837	2,950	3,648	4,106	3,099
		A/S (SL elev./ISA temp.)	1,722	2,887	3,666	4,746	3,687
port		A/S (5,000' elev.@25C)	2,395	3,740	4,779	6,215	4,859
rformance		VMCA	66	77	80	90	86
		VDEC	79	85	97	101	94
		VXSE	92	90	100	105	115
		Vyse	95	105	108	115	121
		Time to Climb (min.)/Altitude	6/FL 010	7/FL 100	17/FL 250	6/FL 100	13/FL 250
Climb		Initial Engine-Out Rate (fpm)	290	270	474	310	682
		al All-Engine Gradient (ft./nm)	1,107	1,500	1,953	910	1,170
	Initial	Engine-Out Gradient (ft./nm)	219	180	259	170	364
		Certificated	24,000	25,000	30,000	30,000	35,000
eilings (ft.)		All-Engine Service	24,000	25,000	30,000	30,000	35,000
		Engine-Out Service	15,420	11,400	19,170	14,550	26,000
		Sea-Level Cabin		 169	11,065 208	184	15,293 256
	Long	- TAS Fuel Flow	180 432	261	332	412	430
	Range	- Altitude	432 FL 010	FL 100	532 FL 260	FL 100	430 FL 350
	nunge	Specific Range	0.417	0.648	0.627	0.447	0.595
ruise		TAS	220	214	270	234	310
	High	Fuel Flow	610	375	612	605	750
	Speed	Altitude	FL 200	FL 100	FL 200	FL 100	FL 260
		Specific Range	0.361	0.571	0.441	0.387	0.413
		Nautical Miles	NP	543	275	427	332
	Max Payload	Average Speed	NP	180	226	179	267
	(w/available fuel)	Trip Fuel	NP	781	648	1,009	866
		Specific Range/Altitude	NP/NP	0.695/FL 100	0.424/FL 270	0.423/FL 100	0.383/FL 33
		Nautical Miles	1,046	837	1,040	1,188	1,413
	Max Fuel	Average Speed	217	179	252	180	291
BAA IFR Ranges	(w/available payload)	Trip Fuel	3,008	1,220	2,069	2,767	2,961
10-nm		Specific Range/Altitude	0.348/FL 100	0.686/FL 100	0.503/FL 270	0.429/FL 100	0.477/FL 330
ernate)		Nautical Miles	1,046	837	997	1,201	1,070
	Full Fuel	Average Speed	217	179	251	180	288
	(w/4 passsengers)	Trip Fuel	3,008	1,220	1,990	2,770	2,286
		Specific Range/Altitude	0.348/FL 100	0.686/FL 100	0.501/FL 270	0.434/FL 100	0.468/FL 330
		Nautical Miles Average Speed	1,051 218	837 179	1,059 255	1,228 176	1,430 293
	Ferry	Average Speed Trip Fuel	3,008	1,220	2,077	2,787	2,962
		Specific Range/Altitude	0.349/FL 100	0.686/FL 100	0.510/FL 260	0.441/FL 100	0.483/FL 33
		Runway	3,163	1,247	3,168	3,815	3,496
		Flight Time	1+26	1+35	1+14	1+21	1+03
	300 nm	Fuel Used	943	419	747	807	868
		Specific Range/Altitude	0.318/FL 100	0.716/FL 100	0.402/FL 210	0.372/FL 100	0.346/FL 25
		Runway	1,289	1,558	3,369	4,075	3,579
ssions		Flight Time	2+22	3+18	2+22	2+39	2+03
bassengers)	600 nm	Fuel Used	1,773	866	1,351	1,609	1,493
		Specific Range/Altitude	0.338/FL 100	0.693/FL 100	0.444/FL 230	0.373/FL 100	0.402/FL 29
		Runway	1,565	NP	3,662	NP	3,668
	1.000	Flight Time	4+36	NP	3+56	NP	3+28
	1,000 nm	Fuel Used	2,881	NP	1,996	NP	2,146
		Specific Range/Altitude	0.347/FL 100	NP/NP	0.501/FL 270	NP/NP	0.466/FL 33
		Suggested Base Price	NA	\$2,772,000	\$3,870,700	NA	\$6,084,925
morko			EASA/FAR 23 pending	FAR 23, 1986 B&CA computed data;	CAR 3, 1959/2007 Pro Line 21; STC wt.	SFAR 41C, 1986	FAR 23, 1973/80/2008/
marks		Certification Basis					

### MULTIENGINE TURBOPROPS >12,500-LB. MTOW

Model B&CA Equippe						Beechcraft King Air 350iE
B&CA Equippe			Avanti II P180	Beechcraft King Air 350HW B300	Beechcraft King Air 350i B300	B300ER
	ed Price		\$7,195,000	\$7,330,255	\$7,421,825	\$8,462,225
Character-		Seating Wing Loading	<u>1+7/9</u> 70.3	<u>1+9/14</u> 53.2	<u>1+9/11</u> 48.4	<u>1+9/11</u> 53.2
stics		Power Loading	7.12	7.86	7.14	7.86
5005		Noise (dBA)	75.0	81.5	72.1	81.5
xternal		Length	47.3	46.7	46.7	46.7
imensions		Height	13.0	14.3	14.3	14.3
t.)		Span	46.0	57.9	57.9	57.9
nternal		Length: OA/Net Height	17.5/17.5	19.5/19.5	19.5/19.5	19.5/19.5
imensions	nensions		5.8	4.8	4.8	4.8
t.)		Width: Max/Floor	6.1/3.5	4.5/4.1	4.5/4.1	4.5/4.1
		Engines	2 P&WC	2 P&WC	2 P&WC	2 P&WC
ower		Output (shp each)/Flat Rating	PT6A-66B 850/ISA+28C	PT6A-60A 1,050/ISA+10C	PT6A-60A 1,050/ISA+10C	PT6A-60A 1,050/ISA+10C
	,	Inspection Interval	3,600t	3,600t	3,600t	3,600t
		Max Ramp	12,150	16,600	15,100	16,600
		Max Takeoff	12,100	16,500	15,000	16,500
		Max Landing	11,500	15,675	15,000	15,675
		Zero Fuel	9,800c	13,000c	12,500c	13,000c
		BOW Max Davlagd	8,375	9,340	10,070	10,330
/eights (lb.)		Max Payload Useful Load	1,425 3,775	3,660 7,260	2,430 5,030	2,670 6,270
		Executive Payload	1,400	1,800	1,800	1,800
		Max Fuel	2,802	3,611	3,611	5,192
		Available Payload w/Max Fuel	973	3,649	1,419	1,078
		Available Fuel w/Max Payload	2,350	3,600	2,600	3,600
	Availa	ble Fuel w/Executive Payload	2,375	3,611	3,230	4,470
		Ммо	260	0.58	0.58	0.58
mite		Trans. Alt. FL	202	FL 240 245	FL 210 263	FL 240 245
imits		VMO VA	<u>9.0</u> 3,262	245 182	182	182
		PSI	4,700	6.5	6.5	6.5
		TO (SL elev./ISA temp.)	5,750	4,598	3,300	4,473
		TOFL (5,000' elev.@25C)	7,400	6,003	5,376	7,588
irport		Hot/High WAT Limit	100	14,491	14,416	15,830
erfor-		NBAA IFR Range	106	1,403	1,548	2,080
nance		V2	132	111	109	111
		VREF Landing Distance	140 10/FL 250	104 2,720	<u>100</u> 2,390	104 2,728
		Time to Climb (min.)/Altitude	670	23/FL 250	15/FL 250	18/FL 250
limb	*FAR 25	Initial Engine-Out Rate (fpm)	1,106	274	552	337
		Engine-Out Gradient (ft./nm)	287	172	304	182
		Certificated	41,000	35,000	35,000	35,000
eilings (ft.)		All-Engine Service	39,400	35,000	35,000	35,000
5111160 (10.)		Engine-Out Service	23,800	17,100	21,500	17,100
		Sea-Level Cabin TAS	24,000 318	15,293 232	15,293 235	15,293 238
	Long	Fuel Flow	408	392	362	402
	Range	Altitude	FL 410	FL 330	FL 330	FL 330
Cruise		Specific Range	0.779	0.592	0.649	0.592
ruise		TAS	400	303	312	303
	High	Fuel Flow	792	766	773	764
	Speed	Altitude	FL 310	FL 240	FL 240 0.404	FL 240
		Specific Range Nautical Miles	0.505	0.396 1,255	899	0.397 1,336
	Max Payload	Average Speed	315	258	274	261
	(w/available fuel)	Trip Fuel	1,715	2,840	1,897	2,886
		Specific Range/Altitude	0.624/FL 390	0.442/FL 350	0.474/FL 350	0.463/FL 350
		Nautical Miles	1,450	1,261	1,489	2,239
		Average Speed	311	258	280	268
BAA IFR		Trip Fuel	2,167	2,886	2,951	4,528
anges		Specific Range/Altitude	0.669/FL 410	0.437/FL 350	0.505/FL 350	0.494/FL 350
00-nm		Nautical Miles	1,510	1,395	1,532	2,278
ternate)		Average Speed Trip Fuel	<u>317</u> 2,167	269 2,908	284 2,956	270 4,531
		Specific Range/Altitude	0.697/FL 410	0.480/FL 350	0.518/FL 350	4,531 0.503/FL 350
		Nautical Miles	1,530	1,429	1,560	2,344
		Average Speed	318	276	288	274
		Trip Fuel	2,167	2,920	2,962	4,539
		Specific Range/Altitude	0.706/FL 410	0.489/FL 350	0.527/FL 350	0.516/FL 350
		Runway	2,350	3,432	2,608	3,451
	300 nm	Flight Time	0+53	<u>1+07</u> 967	<u>1+02</u> 883	1+05
		Fuel Used Specific Range/Altitude	688 0.436/FL 310	967 0.310/FL 250	0.340/FL 250	908 0.330/FL 250
		Specific Range/Altitude Runway	0.436/FL 310 2,550	0.310/FL 250 3,568	2,724	0.330/FL 250 3,584
issions		Flight Time	1+44	2+10	2+02	2+07
passen-	600 nm	Fuel Used	1,144	1,581	1,472	1,510
ers)		Specific Range/Altitude	0.524/FL 350	0.380/FL 290	0.408/FL 290	0.397/FL 290
		Runway	2,700	3,720	2,851	3,732
	1,000 nm	Flight Time	3+02	3+39	3+27	3+36
	1,000 mm	Fuel Used	1,603	2,262	2,106	2,174
		Specific Range/Altitude	0.624/FL 390	0.442/FL 330	0.475/FL 330	0.460/FL 330
emarks	<u> </u>	Suggested Base Price Certification Basis	\$7,195,000 FAR 23, 1990 Includes Pro Line 21 avionics; TCAS I, Iridium satcom;	\$7,484,525 FAR 23, 1989/2007 Commuter category; Installation by HBS; Slick Interior available for Special	\$7,384,125 FAR 23, 1989 Commuter category; Pro Line 21; new interior with Rockwell Collins	\$8,424,525 FAR 23, 1989/07 Commute category; Pro Line 21; new interior with Rockwell Collin

### JETS LESS THAN 20,000-LB. MTOW

Manufacture	r			Embraer	Honda Aircraft Co.	
Model	· · · ·			Phenom 100E EMB-500	HondaJet HA-420	
B&CA Equippe	ed Price			\$4,161,000	\$4,500,000	
- a on Eduippe			Seating	1+5/7	1+5/6	
Character-			Wing Loading	52.5	NA	
istics		Noise	Power Loading (EPNdB): TO/Sideline/APR	3.12 70.4/81.4/86.1	NA NA/NA/NA	
External			Length	42.1	42.6	
Dimensions			Height	14.3	14.9	
(ft.)			Span	40.4	39.8	
Internal			Length: OA/Net	11.0/11.0	12.1/12.1	
Dimensions			Height Width: Max/Floor	4.9	4.8 5.0/NA	
(ft.)			Internal: Cu. ft./lb.	5.1/3.6 10/99	5.0/NA NA/NA	
Baggage			External: Cu. ft./lb.	60/418	66/NA	
			Engines	2 P&WC	2 GE Honda	
Power				PW 617F-E	HF-120	
		0	utput (lb. each)/Flat Rating Inspection Interval	1,695/ISA+10C 3,500t	2,050/NA NA	
			Max Ramp	10,626	NA	
			Max Takeoff	10,582	NA	
			Max Landing	9,877	NA	
			Zero Fuel BOW	8,554c 7,220	NA NA	
			Max Payload	1,334	NA	
Weights (lb.)			Useful Load	3,406	NA	
			Executive Payload	1,000	NA	
		Au		2,804 602	NA NA	
			ailable Payload w/ Max Fuel ailable Fuel w/ Max Payload	2,072	NA	
			e Fuel w/Executive Payload	2,406	NA	
			Ммо	0.700	0.720	
Limits			Trans. Alt. FL/VMo	280/275	FL 300/NA	
			PSI TOFL (SL elev./ISA temp.)	8.3 3,123	8.7 NA	
			TOFL (5,000' elev.@25C)	6,609	NA	
Airport			Hot/High Weight Limit	10,582	NA	
Perfor-			NBAA IFR Range	1,071	NA	
nance			V2@SLISA, MTOW	98	NA	
	Lan		<pre>/ref w/4 Pax, NBAA IFR Res</pre>	94 2,466	NA NA	
	Laii	ung Distai	Time to Climb/Altitude	24/FL 370	NA/NA	
Climb		FAI	R 25 Engine-Out Rate (fpm)	560	ŇA	
		FAR 25 Er	ngine-Out Gradient (ft./nm)	298	NA	
	Certificated			41,000	43,000	
Ceilings (ft.)	All-Engine Service _ Engine-Out Service			41,000 24,045	43,000 NA	
			Sea-Level Cabin	21,280	NA	
			TAS	332	NA	
	Long		Fuel Flow	525	NA	
	Range High Speed		Altitude Specific Range	FL 410 0.632	NA NA	
Cruise		(*******	TAS	389	420	
			Fuel Flow	851	NA	
			Altitude	FL 330	FL 300	
			Specific Range	0.457	NA	
			Nautical Miles	701	NA	
			Average Speed Trip Fuel	319 1,411	NA NA	
			Specific Range/Altitude	0.497/FL 410	NA/NA	
NBAA IFR			Nautical Miles	1,181	1,180	
Ranges			Average Speed	326	NA	
FAR Part 23, LOO-nm			Trip Fuel	2,163	NA	
lternate;			Specific Range/Altitude Nautical Miles	0.546/FL 410 1,050	NA/NA NA	
AR Part 25,		sengers	Average Speed	324	NA	
200-nm			Trip Fuel	1,960	NA	
alternate)			Specific Range/Altitude	0.536/FL 410	NA/NA	
			Nautical Miles Average Speed	<u>1,234</u> 325	NA NA	
			Trip Fuel	2,183	NA	
			Specific Range/Altitude	0.565/FL 410	NA/NA	
			Runway	2,722	NA	
			Flight Time	0+55	NA	
			Fuel Used Specific Range/Altitude	741 0.405/FL 390	NA NA/NA	
licciona			Runway	2,860	NA	
Vissions			Flight Time	1+46	NA	
4 passen-			Fuel Used	1,263	NA	
gers)			Specific Range/Altitude	0.475/FL 390	NA/NA	
			Runway Flight Time	3,050 3+05	NA NA	
		n	Flight lime	1,874	NA	
			Specific Range/Altitude	0.534/FL 410	NA/NA	
Remarks		-	Specific Range/Altitude	0.534/FL 410 FAR 23, 2008	NA/NA FAR 23 pending	

### **JETS** LESS THAN 10,000-LB. MTOW

Manufacture		,000-LB. MTOW	Textron Aviation
Model			Citation Mustang
	a d Duite a		CE-510
B&CA Equippe	ed Price	Castian	\$3,465,000
Character-		Seating Wing Loading	<u>1+5/5</u> 41.2
istics		Power Loading	2.96
100100	N	bise (EPNdB): TO/Sideline/APR	73.9/85.0/86.0
External		Length	40.6
Dimensions		Height	13.4
(ft.)		Span	43.2
Internal		Length: OA/Net	9.8/9.8
Dimensions		Height	4.5
(ft.)		Width: Max/Floor	4.6/3.1
Baggage		Internal: Cu. ft./lb.	6/98
		External: Cu. ft./lb.	57/620 2 P&WC
		Engine(s)	PW615F
Power		Output (lb. each)/Flat Rating	1,460/ISA+10C
		Inspection Interval	3,500t
		Max Ramp	8,730
		Max Takeoff	8,645
		Max Landing Zero Fuel	8,000 6,750c
		Zero Fuel BOW	6,750c 5,595
		Max Payload	1,155
Weights (lb.)		Useful Load	3,135
		Executive Payload	1,000
		Max Fuel	2,580
		Available Payload w/Max Fuel	555
		Available Fuel w/Max Payload	1,980
	Ava	ilable Fuel w/Executive Payload	2,135 0.630
Limits		Ммо Trans. Alt. FL/Vмо	FL 271/250
Ennito		PSI	8.3
		TOFL (SL elev./ISA temp.)	3,110
		TOFL (5,000' elev.@25C)	6,600
Airport		Hot/High Weight Limit	8,645
Perfor-		NBAA IFR Range	988
mance		V2@SLISA, MTOW	97
	Landing D	VREF W/4 Pax, NBAA IFR Res. istance w/4 Pax, NBAA IFR Res.	
	Landing D	Time to Climb/Altitude	20/FL 370
Climb		FAR 25 Engine-Out Rate (fpm)	432
	FAR 2	25 Engine-Out Gradient (ft./nm)	267
		Certificated	41,000
Ceilings (ft.)		All-Engine Service	41,000
		Engine-Out Service Sea-Level Cabin	26,900 21,280
		TAS	319
	Long	Fuel Flow	499
	Range	Altitude	FL 390
Cruise		Specific Range	0.639
oraioo		TAS	339
	High	Fuel Flow	609 FL 350
	Speed	Altitude Specific Range	0.557
		Nautical Miles	716
	Max Payload	Average Speed	294
	(w/available fue		1,300
		Specific Range/Altitude	0.551/FL 410
		Nautical Miles	1,159
NBAA IFR	Max Fuel	Average Speed	305
	(w/available paylo		1,948
Ranges		Specific Range/Altitude Nautical Miles	0.595/FL 410 967
(100-nm alternate)	Four Passenger		301
alternate)	(w/available fuel		1,669
		Specific Range/Altitude	0.579/FL 410
		Nautical Miles	1,205
		Average Speed	316
	TENTY		
	Ferry	Trip Fuel	1,965
	Ferry	Specific Range/Altitude	0.613/FL 410
	Ferry	Specific Range/Altitude Runway	0.613/FL 410 2,496
	Ferry 300 nm	Specific Range/Altitude Runway Flight Time	0.613/FL 410 2,496 1+00
		Specific Range/Altitude Runway	0.613/FL 410 2,496 1+00 670
Missions		Specific Range/Altitude Runway Flight Time Fuel Used	0.613/FL 410 2,496 1+00
Missions (A passen	300 nm	Specific Range/Altitude Runway Flight Time Fuel Used Specific Range/Altitude	0.613/FL 410 2,496 1+00 670 0.448/FL 370
(4 passen-		Specific Range/Altitude Runway Flight Time Fuel Used Specific Range/Altitude Runway Flight Time Fuel Used	0.613/FL 410 2,496 1+00 670 0.448/FL 370 2,695 1+56 1,134
	300 nm	Specific Range/Altitude Runway Flight Time Fuel Used Specific Range/Altitude Runway Flight Time Fuel Used Specific Range/Altitude	0.613/FL 410 2,496 1+00 670 0.448/FL 370 2,695 1+56 1,134 0.529/FL 390
(4 passen-	300 nm	Specific Range/Altitude Runway Flight Time Fuel Used Specific Range/Altitude Runway Flight Time Fuel Used Specific Range/Altitude Runway	0.613/FL 410 2,496 1+00 670 0.448/FL 370 2,695 1+56 1,134 0.529/FL 390 3,109
(4 passen-	300 nm	Specific Range/Altitude Runway Flight Time Fuel Used Specific Range/Altitude Runway Flight Time Fuel Used Specific Range/Altitude Runway Flight Time	0.613/FL 410 2,496 1+00 670 2,695 1+56 1,134 0.529/FL 390 3,109 3+19
(4 passen-	300 nm 600 nm	Specific Range/Altitude Runway Flight Time Fluel Used Specific Range/Altitude Runway Flight Time Fuel Used Specific Range/Altitude Runway Flight Time Fuel Used	0.613/FL 410 2,496 1+00 670 0.448/FL 370 2,695 1+56 1,134 0.529/FL 390 3,109 3+19 1,717
(4 passen-	300 nm 600 nm	Specific Range/Altitude Runway Flight Time Fuel Used Specific Range/Altitude Runway Flight Time Fuel Used Specific Range/Altitude Runway Flight Time	0.613/FL 410 2,496 1+00 670 0.448/FL 370 2,695 1+56 1,134 0.529/FL 390 3,109 3+19 1,717 0.582/FL 410
(4 passen- gers)	300 nm 600 nm	Specific Range/Altitude Runway Flight Time Fuel Used Specific Range/Altitude Runway Flight Time Fuel Used Specific Range/Altitude Runway Flight Time Fuel Used Specific Range/Altitude	0.613/FL 410 2,496 1+00 670 0.448/FL 370 2,695 1+56 1,134 0.529/FL 390 3,109 3+19 1,717 0.582/FL 410 FAR 23, 2006
(4 passen-	300 nm 600 nm	Specific Range/Altitude Runway Flight Time Fluel Used Specific Range/Altitude Runway Flight Time Fuel Used Specific Range/Altitude Runway Flight Time Fuel Used	0.613/FL 410 2,496 1+00 670 0.448/FL 370 2,695 1+56 1,134 0.529/FL 390 3,109 3+19 1,717 0.582/FL 410

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### JETS LESS THAN 20,000-LB. MTOW

Manufacture	r		Textron Aviation	Syberjet	Textron Aviation	Textron Aviation	Embraer	Textron Aviation
Model			Cessna Citation M2 CE-525	SJ30-2	Citation CJ2+ CE-525A	Citation CJ3+ CE-525B	Phenom 300 EMB-505	Citation CJ4 CE-525C
B&CA Equippe	ed Price		\$4,655,000	\$7,250,000	\$7,270,000	\$8,435,000	\$8,955,000	\$9,395,000
Balan Equipp		Seating	1+7/7	1+5/6	1+8/9	1+8/9	1+7/10	2+8/9
Character-		Wing Loading	44.6	73.2	47.4	47.2	58.6	51.8
istics		Power Loading	2.72	3.03	2.51	2.46	2.67	2.36
	Noise	(EPNdB): TO/Sideline/APR	73.2/85.9/88.5	78.5/86.2/91.8	75.5/86.1/89.7	74.0/88.7/88.6	69.9/88.8/88.5	75.4/92.8/89.5
External		Length	42.6	46.8	47.7	51.2	51.2	53.3
Dimensions		Height	13.9	14.2	49.8	15.2	<u>16.7</u> 52.2	15.3
(ft.)		Span	47.3	42.3		53.3		50.8
Internal		Length: OA/Net Height	<u>11.0/11.0</u> 4.8	<u>12.5/12.5</u> 4.4	<u>13.6/13.6</u> 4.8	<u>15.7/15.7</u> 4.8	<u>17.2/17.2</u> 4.9	17.3/17.3 4.8
Dimensions		Width: Max/Floor	4.8/3.1	4.8/2.8	4.8/3.1	4.8/3.1	5.1/3.6	4.8/3.3
(ft.)			4.0/3.1		4.6/3.1	4.8/3.1		
Baggage		Internal: Cu. ft./lb. External: Cu. ft./lb.	46/725	6/100 53/500	65/1,000	65/1,000	10/77 74/573	6/40 71/1,000
			2 Wms Intl	2 Wms Intl	2 Wms Intl	2 Wms Intl	2 P&WC	2 Wms Intl
Deurer		Engines	FJ44-1AP-21	FJ44-2A	FJ44-3A-24	FJ44-3A	PW 535E	FJ44-4A
Power	0	utput (lb. each)/Flat Rating	1,965/ISA+7C	2,300/ISA+8C	2,490/ISA+7C	2,820/ISA+11C	3,360/ISA+15C	3,621/ISA+11C
		Inspection Interval	3,500t	3,500t	4,000t	4,000t	5,000t	5,000t
		Max Ramp	10,800	14,050	12,625	14,070	18,078	17,230
		Max Takeoff	10,700	13,950	12,500	13,870	17,968	17,110
		Max Landing	9,900	12,725	11,525	12,750	16,865	15,660
		Zero Fuel	8,400c	10,500c	9,700c	10,510c	13,999c	12,500c
		BOW .	7,000	8,917	8,030	8,580	11,583	10,460 2,040
Weights (lb.)		Max Payload Useful Load	3,800	1,583 5,133	<u>1,670</u> 4,595	1,930 5,490	2,416 6,495	6,770
		Executive Payload	1,400	1,000	1,600	1,600	1,400	1,600
		Max Fuel	3,309	4,850	3,930	4,710	5,353	5,828
	Av	ailable Payload w/Max Fuel	491	283	665	780	1,142	942
		ailable Fuel w/Max Payload	2,400	3,550	2,925	3,560	4,079	4,730
		e Fuel w/Executive Payload	2,400	4,133	2,995	3,890	5,095	5,170
		Ммо	0.710	0.830	0.737	0.737	0.780	0.770
Limits		Trans. Alt. FL/VMO	FL 305/263	FL 295/320	FL 291/278	FL 293/278	FL 263/320	FL 279/305
	PSI		8.5	12.0	8.9	8.9	9.4	9.0
		TOFL (SL elev./ISA temp.)	3,210	3,939	3,360	3,180	3,138	3,190
Airport		TOFL (5,000' elev.@25C)	5,580	8,784	5,180	4,750	5,114	5,021
Airport Perfor-		Hot/High Weight Limit	10,700 1,198	13,125	12,500 1,531	13,870 1,715	17,968 2,019	16,968
		NBAA IFR Range V2@SL ISA, MTOW	111	1,915 112	1,551	114	112	1,942 117
mance	N N	VREF W/4 Pax, NBAA IFR Res.		104	102	99	104	99
	Landing Distance w/4 Pax, NBAA IT R Res.		101 2,340	2,657	2,658	2,424	2,220	2,281
	Landing blotta	Time to Climb/Altitude	18/FL 370	16/FL 370	15/FL 370	15/FL 370	14/FL 370	14/FL 370
Climb	FAI	R 25 Engine-Out Rate (fpm)	618	312	611	808	911	839
	FAR 25 Engine-Out Gradient (ft./nm)		334	167	316	425	462	430
		Certificated	41,000	49,000	45,000	45,000	45,000	45,000
Ceilings (ft.)		All-Engine Service	41,000	44,000	45,000	45,000	45,000	45,000
oonings (it.)		Engine-Out Service	26,800	25,800	23,800	26,250	30,137	28,200
		Sea-Level Cabin	22,027	41,000	23,586	23,586	25,560	23,984
	TAS		323 516	436	357	352	383	377
		Long Fuel Flow		684 FL 450	591 FL 450	624 FL 450	757 FL 450	812 FL 450
	Range	. Altitude Specific Range	FL 410 0.626	0.637	0.604	0.564	0.506	0.464
Cruise		TAS	401	475	413	415	444	442
	High	Fuel Flow	920	1,188	1,096	1,197	1,312	1,470
	Speed	Altitude	FL 350	FL 360	FL 350	FL 350	FL 350	FL 370
		Specific Range	0.436	0.400	0.377	0.347	0.338	0.301
		Nautical Miles	812	1,635	993	1,172	1,247	1,425
	Max Payload	Average Speed	361	402	368	368	397	407
	(w/available fuel)	Trip Fuel	1,706	2,908	2,071	2,552	3,109	3,753
		Specific Range/Altitude	0.476/FL 410	0.562/FL 470	0.479/FL 450	0.459/FL 450	0.401/FL 450	0.380/FL 450
NBAA IFR		Nautical Miles	1,369	2,598	1,610	1,869	1,877	1,913
Ranges	Max Fuel	Average Speed	373	410	379	378	409	413
(FAR Part 23,	(w/available payload)	Trip Fuel	2,676	4,241	3,152	3,850	4,416	4,904
100-nm		Specific Range/Altitude	0.512/FL 410	0.613/FL 490	0.511/FL 450	0.485/FL 450	0.425/FL 450	0.390/FL 450
alternate;		Nautical Miles	1,177	2,205	1,509	1,691	1,903	1,919
FAR Part 25,	Four Passengers (w/available fuel)	Average Speed	370	408	377	376	411	414
200-nm	(wavaliable fuer)	Trip Fuel Specific Range/Altitude	2,342 0.503/FL 410	3,713 0.594/FL 490	2,975 0.507/FL 450	3,518 0.481/FL 450	4,447 0.428/FL 450	4,911 0.391/FL 450
alternate)		Nautical Miles	1,398	2,667	1,646	1,890	1,944	1,942
		Average Speed	378	411	384	381	418	415
	Ferry	Trip Fuel	2,704	4,246	3,177	3,865	4,473	4,911
		Specific Range/Altitude	0.517/FL 410	0.628/FL 490	0.518/FL 450	0.489/FL 450	0.435/FL 450	0.395/FL 450
		Runway	2,626	2,822	2,479	2,604	2,613	2,433
	200	Flight Time	0+52	0+45	0+49	0+49	0+47	0+46
	300 nm	Fuel Used	804	846	899	972	1,058	1,089
		Specific Range/Altitude	0.373/FL 370	0.355/FL 410	0.334/FL 370	0.309/FL 370	0.284/FL 390	0.275/FL 390
Missions		Runway	2,694	3,025	2,694	2,617	2,747	2,449
(4 passen-	600 nm	Flight Time	1+38	1+26	1+35	1+35	1+29	1+27
	000 mm	Fuel Used	1,362	1,313	1,460	1,576	1,735	1,868
gers)		Specific Range/Altitude	0.441/FL 390	0.457/FL 450	0.411/FL 410	0.381/FL 410	0.346/FL 410	0.321/FL 410
		Runway	3,006	3,336	2,994	2,786	2,808	2,530
	1,000 nm	Flight Time	2+43	2+21	2+36	2+37	2+26	2+24
		Fuel Used . Specific Range/Altitude	2,018 0.496/FL 410	1,980 0.505/FL 450	2,162 0.463/FL 430	2,324 0.430/FL 430	2,471 0.405/FL 450	2,829 0.353/FL 430
		opeone nange/Altitude	0.430/12 410	· · · · · · · · · · · · · · · · · · ·	0.703/12 430	FAR 23 Commuter	·	
Remarks		Certification Basis	FAR 23, 2013	FAR 23	FAR 23, 2000/05	category, 2004/2014;	FAR 23 Commuter	FAR 23 Commuter
			.,	Commuter category	.,,	Garmin G3000.	category, 2009	category, 2010

Manufacturer Model			Bombardier Learjet 70 Model 45	Textron Aviation Citation XLS+ CE-560XL	Bombardier Learjet 75 Model 45	Bombardier Learjet 60XR Model 60	Gulfstream Aero. Gulfstream 150 G150	Textron Aviation Citation Latitude CE-680A
B&CA Equippe	d Price		\$11,300,000	\$12,920,000	\$13,800,000	\$14,674,000*	\$15,700,000	\$16,250,000
		Seating	2+6/7	2+9/12	2+8/9	2+7/9	2+7/9	2+9/10
Character-		Wing Loading	69.0	54.6	69.0	88.8	82.3	NA
stics		Power Loading	2.79	2.45	2.79	2.55	2.95	NA
	Nois	e (EPNdB): TO/Sideline/APR	75.5/85.0/93.4	72.2/86.8/92.8	75.5/85.0/93.4	70.8/83.1/87.7	80.7/91.2/91.9	NA/NA/NA
xternal		Length	56.0 14.0	52.5 17.2	58.0 14.0	58.7 14.6	56.8 19.1	62.3 20.9
imensions		Height						
t.)		Span	50.9	56.3	50.9	43.8	55.6	72.3
nternal		Length: OA/Net	17.7/17.7	18.5/18.5	19.8/19.8	17.7/15.8	17.7/17.7	28.1/21.8
Dimensions		Height	4.9	5.7	4.9	5.7	5.8	6.0
ft.)		Width: Max/Floor	5.1/3.2	5.5/3.9	5.1/3.2	5.9/3.8	5.8/4.7	6.4/4.1
aggage		Internal: Cu. ft./lb.	15/150	10/100	15/150	24/350	25/NA	26/NA
ueeuec		External: Cu. ft./lb.	50/500	80/700	50/500	24/300	55/1,100	100/1,000
		Engines	2 Hon	2 P&WC	2 Hon	2 P&WC	2 Hon	2 P&WC
ower		-	TFE731-40BR	PW545C	TFE731-40BR	PW305A	TFE731-40AR-200G	PW306D
		Output (Ib. each)/Flat Rating	3,850/ISA+23C	4,119/ISA+10C	3,850/ISA+23C	4,600/ISA+17C	4,420/ISA+13C	5,852/ISA+150
		Inspection Interval	6,000t	5,000t	6,000t	6,000t	6,000c	6,000t
		Max Ramp	21,750	20,400	21,750	23,750	26,250 26,100	NA
		Max Takeoff	21,500	20,200	21,500	23,500		NA
		Max Landing Zero Fuel	19,200 16,000c	18,700 15,100c	19,200 16,000c	19,500 17,000c	21,700 17,500c	NA
·								
		BOW Max Payload	<u>13,715</u> 2,285	12,860 2,240	13,890 2,110	14,896 2,104	15,200 2,300	NA
/eights (lb.)		Useful Load	8,035	7,540	7,860	8,854	11,050	NA
		Executive Payload	1,200	1,800	1,600	1,400	1,400	1,800
		Max Fuel	6,062	6,740	6,062	7,910	10,300	
	,	Available Payload w/Max Fuel	1,973	800	1,798	944	750	1,000
		Available Fuel w/Max Payload	5,750	5,300	5,750	6,750	8,750	
		ble Fuel w/Executive Payload	6,062	5,740	6,062	7,454	9,650	NA
	, wana	MMO	0.810	0.750	0.810	0.810	0.850	0.800
imits		Trans. Alt. FL/VMO	FL 270/330	FL 265/305	FL 270/330	FL 270/330	FL 300/330	FL 298/305
		PSI	9.4	9.3	9.4	9.4	8.8	9.7
		TOFL (SL elev./ISA temp.)	4,440	3,560	4,440	5,450	5,012	4,030
		TOFL (5,000' elev.@25C)	5,091	5,430	5,186	8,540	8,120	5,000
irport		Hot/High Weight Limit	20,447	20,200	20,622	23,402p	26,100	NA
erfor-		NBAA IFR Range	1,934	1,740	1,912	2,275	2,988	NA
ance		V2@SLISA, MTOW	125	118	125	147	131	NA
nanoo	VREF W/4 Pax, NBAA IFR Res.		112	106	113	131	115	NA
	Landing Dist	ance w/4 Pax, NBAA IFR Res.	2,332	2,740	2,347	3,049	2,442	2,250
	Time to Climb/Altitude FAR 25 Engine-Out Rate (fpm)		15/FL 370	15/FL 370	15/FL 370	13/FL 370	17/FL 370	15/FL 370
limb			430	765	430	718	438	NA
	FAR 25 Engine-Out Gradient (ft./nm)		207	389	207	293	201	NA
		Certificated	51,000	45,000	51,000	51,000	45,000	45,000
cilingo (ft.)		All-Engine Service	44,700	45,000	44,700	42,400	42,400	43,000
eilings (ft.)		Engine-Out Service	27,900	28,600	27,900	24,300	26,400	NA
	Sea-Level Cabin		25,700	25,230	25,700	25,700	23,000	NA
		TAS	437	353	437	423	430	NA
	Long	Fuel Flow	961	865	970	1,128	1,184	NA
	Range	Altitude	FL 470	FL 450	FL 470	FL 430	FL 430	NA
Cruise		Specific Range	0.455	0.408	0.451	0.375	0.363	NA
		TAS	453	431	452	446	475	396
	High	Fuel Flow	1,082	1,238	1,080	1,288	1,938	1,750
	Speed	Altitude	FL 470	FL 410	FL 470	FL 430	FL 350	FL 390
		Specific Range	0.419	0.348	0.419	0.346	0.245	0.226
		Nautical Miles	1,595	1,150	1,595	1,742	2,335	NA
		Average Speed	422	383	422	413	415	NA
		Trip Fuel	4,285	3,663	4,285	5,255	7,265	NA
		Specific Range/Altitude	0.372/FL 470	0.314/FL 450	0.372/FL 470	0.331/FL 410	0.321/FL 450	NA/NA
		Nautical Miles	1,747	1,745	1,747	2,243	3,011	2,500
BAA IFR		Average Speed	424	395	424	415	418	NA
			4,616	5,236	4,616	6,486	8,903	NA NA (5) 450
anges		Specific Range/Altitude	0.378/FL 470	0.333/FL 450	0.378/FL 470	0.346/FL 410	0.338/FL 450	NA/FL 450
00-nm		Nautical Miles	1,934	1,719	1,912	2,292	2,988	NA
ternate)		Average Speed	425	395	425	414	418	NA
			4,803	5,168	4,793	6,594	8,850	NA
		Specific Range/Altitude	0.403/FL 470	0.333/FL 450	0.399/FL 470	0.348/FL 410	0.338/FL 450	NA/NA
		Nautical Miles	2,039	1,785	2,015	2,398	3,122	NA
		Average Speed	425 4,848	403 5,268	425 4,838	412 6,642	419 8,945	NA NA
		Trip Fuel Specific Pange (Altitude			4,838 0.416/FL 490			
		Specific Range/Altitude	0.421/FL 490	0.339/FL 450		0.361/FL 410	0.349/FL 450	NA/NA
		Runway	3,595	2,734	3,606	3,308	3,623	NA
	300 nm	Flight Time	0+45	0+46	0+45	0+45	0+50	NA
		Fuel Used	1,097	1,246	1,104	1,109	1,230	NA NA (NA
		Specific Range/Altitude	0.273/FL 450	0.241/FL 390	0.272/FL 450	0.271/FL 450	0.244/FL 450	NA/NA
lissions		Runway	3,642	2,758	3,654	3,552	3,783	NA
passen-	600 nm	Flight Time	1+24	1+29	1+24	1+26	1+32	NA
ers)		Fuel Used	1,885	2,094	1,898	1,909	1,974	NA NA (NA
		Specific Range/Altitude	0.318/FL 450	0.287/FL 410	0.316/FL 450	0.314/FL 450	0.304/FL 450	NA/NA
		Runway Elight Timo	3,708	3,028	3,720	3,928	3,971	NA
	1,000 nm	Flight Time	2+16	2+26	2+17	2+21	2+28	NA
		Fuel Used	2,981	3,211	2,995	3,011	2,998	NA
		Specific Range/Altitude	0.335/FL 450	0.311/FL 430	0.334/FL 450	0.332/FL 450	0.334/FL 450	NA/NA
emarks		Certification Basis	FAR 25, EASA CS 25	FAR 25, 2008	FAR 25, EASA CS 25	FAR 25, 1981/92/2006 *2013 dollars.	FAR 25 A108, 2005	FAR 25 pending All data preliminar Garmin G5000.

Manufacturer	·		<b>Textron Aviation</b>	Embraer	Textron Aviation	Gulfstream Aero.	Bombardier	Embraer
Model			Citation Sovereign+	Legacy 500	Cessna Citation X+	Gulfstream 280	Challenger 300	Legacy 600
B&CA Equippe	d Price		CE-680 \$18,110,000	EMB-550 \$19,995,000	CE-750 \$23,500,000	<b>G280</b> \$24,500,000	BD-100-1A10 \$24.860.000	EMB-135BJ \$26,000,000
Data Lquippe		Seating	2+9/12	2+8/12	2+9/12	2+10/19	2+9/11	2+13/14
Character-		Wing Loading	56.7	NA	NA	80.0	74.4	90.0
istics		Power Loading	2.60	NA	2.60	2.60	2.85	3.12
	Nois	e (EPNdB): TO/Sideline/APR	71.9/87.2/88.1	NA/NA/NA	NA/NA/NA	75.2/89.5/90.5	75.5/87.6/89.6	79.7/86.8/91.3
External		Length	63.5	68.1	73.6	66.8	68.7	86.4
Dimensions		Height	20.3	21.2	19.2 69.2	21.3	20.0	22.2
(ft.)		Span	72.3	66.4		63.0	63.8	69.5
Internal		Length: OA/Net Height	25.3/25.3 5.7	27.5/24.6 6.0	25.2/25.2 5.7	32.3/25.8 6.3	28.6/19.6 6.1	49.8/42.4 6.0
Dimensions		Width: Max/Floor	5.5/3.9	6.8/4.7	5.5/3.9	7.2/5.7	7.2/5.1	6.9/5.2
(ft.)		Internal: Cu. ft./lb.	35/435	40/NA	22/NA	154/1,980	106/750	286/1,441
Baggage		External: Cu. ft./lb.	100/1,000	110/NA	82/775	/	/	/
			2 P&WC	2 Hon	2 RR	2 Hon	2 Hon	2 RR
Power		Engines	PW306D	HTF7500E	AE3007C2	HTF7250G	HTF 7000	AE 3007 A1E
		Output (lb. each)/Flat Rating	5,907/ISA+16C	6,540/ISA+15C	7,034/ISA+15C	7,624/ISA+17C	6,826/ISA+15C	7,953/ISA+22C
		Inspection Interval	6,000t 31,025	OC NA	4,500t* 36,900	0C 39,750	0C 39,000	0C 49,758
		Max Ramp Max Takeoff	30,775	NA	36,600	39,600	38,850	49,758
		Max Landing	27,575	NA	32,000	32,700	33,750	40,785
		Zero Fuel	21,000c	NA	24,978c	28,200c	27,200c	35,274c
		BOW	18,330	NA	22,464	24,150	23,850*	30,081
Weights (lb.)		Max Payload	2,670	2,800	2,514	4,050	3,350	5,193
		Useful Load	12,695	NA	14,436	15,600	15,150	19,677
		Executive Payload	1,800	1,600	1,800	2,000	1,800	2,600
		Max Fuel	11,390	NA 1.600	12,931	14,600	14,043	18,170
		wailable Payload w/Max Fuel	1,305 10.025	1,600 NA	1,505 11,922	1,000 11.550	1,107 11,800	1,507 14,484
		ble Fuel w/Executive Payload	10,025	NA	11,922 12,636	11,550	13,350	14,484
	Availa	MM0	0.800	0.830	0.935	0.850	0.830	0.800
Limits		Trans. Alt. FL/VMo	FL 298/305	FL 295/320	FL 307/350	FL 280/340	FL 290/320	FL 276/320
		PSI	9.3	9.6	9.3	9.2	8.8	8.4
		TOFL (SL elev./ISA temp.)	3,530	4,600	5,280	4,750	4,810	5,614
•		TOFL (5,000' elev.@25C)	4,795	NA	7,300	7,320	6,860	7,604
Airport		Hot/High Weight Limit	30,345	NA	35,400p	39,600	38,545	49,604
Perfor-		NBAA IFR Range V2@SL ISA, MTOW	3,085	NA NA	3,164 NA	3,600 137	3,250 130	3,453 139
mance		VREF W/4 Pax, NBAA IFR Res.	96	NA	NA	117	113	113
		ance w/4 Pax, NBAA IFR Res.	2,149	NA	2,735	2,642	2,290	2,301
	8	Time to Climb/Altitude	13/FL 370	14/FL 370	14/FL 370	14/FL 370	14/FL 370	21/FL 370
Climb	FAR 25 Engine-Out Rate (fpm)		735	NA	NA	845	474	630
	FAR 25 Engine-Out Gradient (ft./nm)		377	NA	NA	371	219	272
	Certificated		47,000	45,000	51,000	45,000	45,000	41,000
Ceilings (ft.)		All-Engine Service	45,000	44,000	45,000	45,000	44,000	40,900
0,		Engine-Out Service Sea-Level Cabin	29,740 25,230	NA 26,520	NA 25,230	27,500 25,000	27,800 23,100	23,276 21,650
		TAS	368	421	470	459	459	424
	Long	Fuel Flow	1,059	NA	1,427	1,478	1,584	1,879
	Range	Altitude	FL 450	NA	FL 470	FL 450	FL 450	FL 410
Cruise		Specific Range	0.347	NA	0.329	0.311	0.290	0.226
Ciuise		TAS	448	467	516	482	470	455
	High	Fuel Flow	1,756	NA	2,329	1,910	1,809	2,545
	Speed	Altitude _	FL 390 0.255	NA NA	FL 410 0.222	FL 430 0.252	FL 430 0.260	FL 370 0.179
		Specific Range Nautical Miles	2,484	NA	2,813	2,544	2,522	2,417
		Average Speed	396	NA	463	434	445	414
		Trip Fuel	8,170	NA	9,959	9,591	9,889	12,242
		Specific Range/Altitude	0.304/FL 470	NA/NA	0.282/FL 490	0.265/FL 450	0.255/FL 450	0.197/FL 410
		Nautical Miles	3,025	NA	3,229	3,590	3,229	3,376
		Average Speed	400	NA	464	441	448	407
NBAA IFR			9,661	NA	11,122	12,657	12,203	16,065
Ranges		Specific Range/Altitude	0.313/FL 470	NA/NA	0.290/FL 490	0.284/FL 450	0.265/FL 450	0.210/FL 410
200-nm		Nautical Miles	3,061	3,000	3,295	3,600	3,250	3,430
alternate)		Average Speed Trip Fuel	<u>401</u> 9,676	NA NA	464 11,151	442 12,761	448 12.213	406 16,094
		Specific Range/Altitude	9,676 0.316/FL 470	NA NA/FL 450	0.295/FL 490	0.282/FL 450	0.266/FL 450	0.213/FL 410
		Nautical Miles	3,130	NA/PL 450	3,380	3,686	3,310	3,485
		Average Speed	405	NA	465	442	448	402
		Trip Fuel	9,705	NA	11,188	12,792	12,239	16,122
		Specific Range/Altitude	0.323/FL 470	NA/NA	0.302/FL 490	0.288/FL 450	0.270/FL 450	0.216/FL 410
		Runway	2,592	NA	3,740	2,957	3,370	3,522
	300 nm	Flight Time	0+45	NA	0+40	0+48	0+47	0+48
		Fuel Used	1,507	NA	1,847	1,524	1,578	1,894
		Specific Range/Altitude	0.199/FL 390 2.600	NA/NA	0.162/FL 370 3 790	0.197/FL 450	0.190/FL 450	0.158/FL 410 3 716
Missions		Runway _ Flight Time	2,600 1+26	NA 1+31	3,790 1+15	2,997 1+27	3,420 1+26	3,716 1+37
4 passen-	600 nm	Fuel Used	2,406	2,474	2,915	2,443	2,568	3,044
gers)		Specific Range/Altitude	0.249/FL 430	0.243/FL 450	0.206/FL 430	0.246/FL 450	0.234/FL 450	0.197/FL 410
		Runway	2,650	NA	3,860	3,136	3,498	3,789
	1 000 pm	Flight Time	2+21	NA	2+02	2+19	2+18	2+36
	1,000 nm	Fuel Used	3,753	NA	4,588	3,692	3,910	4,731
		Specific Range/Altitude	0.266/FL 430	NA/NA	0.218/FL 430	0.271/FL 450	0.256/FL 450	0.211/FL 410
Remarks		Certification Basis	FAR 25, 2013 Garmin G5000.	FAR 25, EASA CS 25, pending All data preliminary.	FAR 25, 2014 Garmin G5000; *Engine flight hour inspection interval.	FAR 25, 2012 and EASA CS 25, 2013	FAR 25 A 98 and JAR 25 Chg 15 *B&CA Operators Survey BOW 24,120 lb.	FAR 25, 2002

Manufacturer Model		Bombardier Challenger 350 BD-100-1A10	Dassault Falcon 2000S Falcon 2000EX	Bombardier Challenger 605 CL-600-2B16	Embraer Legacy 650 EMB-135BJ*	Bombardier Challenger 850 CL-600-2B19	
B&CA Equipped Price			\$26,540,000	\$27,700,000	\$31,100,000	\$31,600,000	\$31,981,000*
COA Equippe		Seating	2+9/11	2+10/19	3+10/19	2+13/14	2+14/15
haracter-		Wing Loading	77.8	77.7	107.1	97.2	101.8
tics		Power Loading	2.77	2.93	2.76	2.97	3.04
	Nois	e (EPNdB): TO/Sideline/APR	75.5/87.6/89.6	75.1/91.8/90.5	81.2/86.2/90.3	78.0/86.9/91.7	78.8/82.4/92.1
ternal		Length	68.7	66.3	68.4	86.4	87.8
mensions		Height	20.0	23.2	20.7	21.8	20.4
.)		Span	69.0	70.2	64.3	69.5	69.6
ernal		Length: OA/Net	28.6/19.6	31.2/26.2	28.3/25.5	49.8/42.4	48.4/40.2
mensions		Height	6.1	6.2	6.1	6.0	6.0
.)		Width: Max/Floor	7.2/5.1	7.7/6.3	7.9/6.9	6.9/5.2	7.9/6.9
		Internal: Cu. ft./lb.	106/750	131/1,600	115/900	286/1,441	147/900
ggage		External: Cu. ft./lb.	_/_	_/_			_/_
		,	2 Hon	2 P&WC	2 GE	2 RR	2 GE
		Engines	HTF 7350	PW308C	CF34-3B	AE 3007A2	CF34-3B1
wer		Output (Ib. each)/Flat Rating	7,323/ISA+15C	7,000/ISA+15C	8,729/ISA+15C	9,020/ISA+15C	8,729/ISA+8C
		Inspection Interval	00	7,000c	00	00	00
		Max Ramp	40,750	41,200	48,300	53,727	53,250
		Max Takeoff	40,600	41,000	48,200	53,572	53,000
		Max Landing	34,150	39,300	38,000	44,092	47,000
		Zero Fuel	28,200c	29,700c	32,000c	36,156c	44,000c
		BOW	24,800	24,750	27,150	31,217	34,618
eights (lb.)		Max Payload Useful Load	<u>3,400</u> 15,950	4,950 16,450	4,850 21,150	4,939	9,382 18.632
		Executive Payload	15,950	2,000	21,150 2,000	22,510 2,600	2,800
		Max Fuel	14,043	14,600	19,852	20,600	18,274
	Δ	vailable Payload w/Max Fuel	1,907	1,850	1,298	1,910	358
		vailable Fuel w/Max Payload	12,550	11,500	16,300	17,571	9,250
		ble Fuel w/Executive Payload	14,043	14,450	19,150	19,910	15,832
		MM0	0.830	0.862	0.850	0.800	0.850
nits		Trans. Alt. FL/VMo	FL 290/320	FL 250/370	FL 222/348	FL 276/320	FL 254/335
		PSI	8.8	9.3	8.8	8.4	8.6
		TOFL (SL elev./ISA temp.)	4,835	4,325	5,840	5,741	6,305
		TOFL (5,000' elev.@25C)	6,890	6,050	9,192	7,979	11,332
rport		Hot/High Weight Limit	39,493	39,950	47,702	53,572	53,000p
rfor-		NBAA IFR Range	3,250	3,555	4,038	3,953	2,986
ance	V2@SL ISA, MTOW Vref w/4 Pax, NBAA IFR Res.		NA	123	147	144	146
			NA	106	117	115	126
	Landing Distance w/4 Pax, NBAA IFR Res.		2,320	2,300	2,360	2,346	2,475
	Time to Climb/Altitude		14/FL 370	14/FL 370	21/FL 370	21/FL 370	32/FL 370
imb	FAR 25 Engine-Out Rate (fpm)		NA	535	581	633	443
	FAR 25 I	Engine-Out Gradient (ft./nm)	NA	261	237	259	182
		Certificated	45,000	47,000 43,700	41,000	41,000	41,000
eilings (ft.)	All-Engine Service _ Engine-Out Service _ Sea-Level Cabin		44,000 NA	26,150	38,250 20,000	41,000 23,128	37,760 19,370
			23,100	25,300	20,000	23,128 21,650	21,100
		TAS	459	437	424	425	424
	Long	Fuel Flow	1,574	1,455	1,828	1,901	2,061
	Range		FL 450	FL 450	FL 410	FL 410	FL 390
		Specific Range	0.292	0.300	0.232	0.224	0.206
ruise		TAS	470	482	470	459	459
	High	Fuel Flow	1,849	2,280	2,443	2,570	2,393
	Speed	Altitude	FL 430	FL 390	FL 370	FL 370	FL 370
		Specific Range	0.254	0.211	0.192	0.179	0.192
		Nautical Miles	2,702	2,450	3,010	3,076	1,116
		Average Speed	448	426	416	417	402
		Trip Fuel	10,633	9,640	14,256	15,238	6,759
		Specific Range/Altitude	0.254/FL 450	0.254/FL 450	0.211/FL 410	0.202/FL 410	0.165/FL 370
		Nautical Miles	3,166	3,445	3,973	3,839	3,089
BAA IFR		Average Speed	438	429	418	417	416
			12,171	12,740	17,939	18,380	16,111
nges		Specific Range/Altitude	0.260/FL 450	0.270/FL 470	0.221/FL 410	0.209/FL 410	0.192/FL 390
)0-nm		Nautical Miles	3,250	3,540	4,047	3,919	2,986
ernate)		Average Speed	449	431	418	415	416
		Trip Fuel	12,203	12,740	18,105	18,422 0.213/FL 410	15,652 0.101/EL 200
		Specific Range/Altitude	0.266/FL 450	0.278/FL 470	0.224/FL 410 4,121		0.191/FL 390 3 110
		Nautical Miles	3,304 451	3,615 430	4,121 418	3,980 414	3,110 416
		Average Speed Trip Fuel	12,225	12,740	18,134	18,450	16,124
		Specific Range/Altitude	0.270/FL 450	0.284/FL 470	0.227/FL 410	0.216/FL 410	0.193/FL 390
		Runway	3,570	2,900	3,389	3,346	3,858
		Flight Time	0+47	0+48	0+47	0+49	0+48
	300 nm	Fuel Used	1,583	1,525	1,593	1,773	2,143
		Specific Range/Altitude	0.190/FL 450	0.197/FL 450	0.188/FL 390	0.169/FL 410	0.140/FL 390
opione		Runway	3,620	2,905	3,428	3,518	4,109
ssions		Flight Time	1+26	1+28	1+25	1+34	1+28
passen-	600 nm	Fuel Used	2,577	2,465	3,065	3,146	3,603
rs)		Specific Range/Altitude	0.233/FL 450	0.243/FL 470	0.196/FL 390	0.191/FL 410	0.167/FL 390
		Runway	3,680	3,050	3,490	3,573	4,474
	1 000	Flight Time	2+18	2+21	2+16	2+33	2+21
	1,000 nm	Fuel Used	3,925	3,755	5,097	4,815	5,648
		Specific Range/Altitude	0.255/FL 450	0.266/FL 470	0.196/FL 390	0.208/FL 410	0.177/FL 390
marks		Certification Basis	FAR 25 A 98 and JAR 25 Chg 15	FAR/EASA 25, 2013 EASy II flight deck; 2015 delivery price	FAR 25, 1980/83/87/95/2006 Pro Line 21;	FAR 25, 2011 *Factory modification DCA 145-000-	FAR/JAR 25, 1992 Optional ISA+15C en flat-rating; *2013 do

Manufacturer Model			Dassault Falcon 2000LXS Falcon 2000EX	Gulfstream Aerospace Gulfstream 450 GIV-X	Dassault Falcon 900LX Falcon 900EX	Bombardier Global 5000 BD-700-1A11	Dassault Falcon 7X Falcon 7X
B&CA Equippe	uipped Price		\$32,900,000	\$42,200,000	\$42,200,000	\$50,190,000	\$52,800,000
Lon Equippe		Seating	2+8/19	2+14/19	2+12/19	3+13/19	3+12/19
Character-		Wing Loading	81.2	77.8	92.9	90.5	92.0
stics		Power Loading	3.06	2.69	3.27	3.14	3.64
	Noise (	(EPNdB): TO/Sideline/APR	76.4/91.7/90.5	76.2/89.5/92.3	78.2/90.3/92.1	81.3/88.9/89.7	82.3/90.1/92.6
xternal		Length	66.3	89.3	66.3	96.8	76.1
imensions		Height	23.2	25.2	24.8	25.5	25.7
t.)		Span	70.2	77.8	70.2	94.0	86.0
iternal		Length: OA/Net	31.2/26.2	45.1/37.0	39.0/33.2	45.7/40.7	46.5/39.1
imensions		Height	6.2	6.2	6.2	6.2	6.2
t.)		Width: Max/Floor	7.7/6.3	7.3/5.5	7.7/6.3	7.9/6.5	7.7/6.3
		Internal: Cu. ft./lb.	131/1,600	169/2,000	127/2,866	195/1,000	140/2,004
aggage		External: Cu. ft./lb.	—/—	—/—	—/—	—/—	—/—
		Engines	2 P&WC	2 RR	3 Hon	2 RR	3 P&WC
ower			PW308C	Tay Mk 611-8C	TFE731-60	BR700-710A2-20	PW307A
	0ι	utput (lb. each)/Flat Rating	7,000/ISA+15C	13,850/ISA+15C	5,000/ISA+17C	14,750/ISA+20C	6,402/ISA+17C
		Inspection Interval	7,000c	12,000t or 0C	6,000c	00	7,200c
		Max Ramp	43,000	75,000	49,200	92,750	70,200
		Max Takeoff	42,800	74,600	49,000	92,500	70,000
		Max Landing	39,300	66,000	44,500	78,600	62,400
		Zero Fuel BOW	29,700c	49,000c	30,864c 26,750	58,000c	41,000c
		Max Payload	24,750 4,950	43,200 5,800	4,114	50,861 7,139	36,600 4,400
eights (lb.)		Useful Load	18,250	31,800	22,450	41,889	33,600
		Executive Payload	1,600	2,800	2,400	2,600	2,400
		Max Fuel	16,660	29,281	20,905	38,959	31,940
	Ava	ailable Payload w/Max Fuel	1,590	2,519	1,545	2,930	1,660
		ailable Fuel w/Max Payload	13,300	26,000	18,336	34,750	29,200
		Fuel w/Executive Payload	16,650	29,000	20,050	38,959	31,200
		Ммо	0.862	0.880	0.870	0.890	0.900
Limits		Trans. Alt. FL/VMo	FL 250/370	FL 280/340	FL 250/370	FL 303/340	FL 270/370
		PSI	9.3	9.6	9.6	10.3	10.2
		TOFL (SL elev./ISA temp.)	4,675	5,600	5,360	5,540	5,710
		TOFL (5,000' elev.@25C)	6,800	8,200	7,615	6,798	8,045
rport		Hot/High Weight Limit	42,010	74,600	48,255	88,373	69,140
erfor-		NBAA IFR Range	4,095	4,328	4,710	5,185	5,795
ance	V2@SLISA, MTOW		127	150	134	133	133
	VREF W/4 Pax, NBAA IFR Res.		106	123	111	107	106
	Landing Distance w/4 Pax, NBAA IFR Res.		2,300	2,663	2,432	2,189	2,120
limb	Time to Climb/Altitude FAR 25 Engine-Out Rate (fpm) FAR 25 Engine-Out Gradient (ft./nm)		16/FL 370 464	16/FL 370 712	18/FL 370 703	18/FL 370 704	18/FL 370 615
			219	285	315	318	280
	FAR 25 Engine-Out Gradient (ft./hm) Certificated		47,000	45,000	51,000	51,000	51,000
	All-Engine Service Engine-Out Service Sea-Level Cabin TAS Long Fuel Flow Range Altitude Specific Range		43,700	42,400	40,100	44,600	41,360
eilings (ft.)			26,150	25,000	31,400	20,600	31,560
			25,300	26,700	25,300	30,125	29,200
			437	459	431	470	459
			1,485	2,585	1,670	2,856	2,260
			FL 450	FL 450	FL 430	FL 450	FL 430
Cruise			0.294	0.178	0.258	0.165	0.203
luise		TAS	482	476	474	505	497
	High	Fuel Flow	2,315	3,055	2,230	3,582	3,205
	Speed	Altitude	FL 390	FL 410	FL 390	FL 410	FL 390
		Specific Range	0.208	0.156	0.213	0.141	0.155
		Nautical Miles	2,905	3,549	3,810	4,958	5,000
		Average Speed	428	452	420	463	453
		Trip Fuel	11,475	22,622	16,386	33,418	26,820
		Specific Range/Altitude	0.253/FL 450	0.157/FL 450	0.233/FL 430	0.148/FL 470	0.186/FL 450
		Nautical Miles	3,980	4,216	4,595	5,381	5,670
BAA IFR		Average Speed	431	453	421	463	454
		Trip Fuel	14,835	26,023	18,955	35,695	29,560
anges		Specific Range/Altitude	0.268/FL 470	0.162/FL 450	0.242/FL 430	0.151/FL 470	0.192/FL 470
00-nm		Nautical Miles	4,075 431	4,328 452	4,695 420	5,520 463	5,760 454
ternate)		Average Speed		452 26,087	420	463 35,761	
		Trip Fuel Specific Range/Altitude	14,835 0.275/FL 470	0.166/FL 450	0.248/FL 430	0.154/FL 470	29,560 0.195/FL 470
		Nautical Miles	4,145	4,382	4,765	5,572	5,840
		Average Speed	431	4,382	4,785	463	454
		Trip Fuel	14,835	26,116	18,955	35,786	29,560
		Specific Range/Altitude	0.279/FL 450	0.168/FL 450	0.251/FL 430	0.156/FL 470	0.198/FL 470
		Runway	2,900	3,225	2,880	2,483	2,500
		Flight Time	0+48	0+46	0+47	0+46	0+46
	300 nm	Fuel Used	1,525	2,599	1,595	2,755	2,075
		Specific Range/Altitude	0.197/FL 450	0.115/FL 450	0.188/FL 450	0.109/FL 490	0.145/FL 450
icciono		Runway	2,905	3,258	2,870	2,572	2,515
lissions	C00	Flight Time	1+28	1+25	1+27	1+23	1+25
passen-	600 nm	Fuel Used	2,465	4,113	2,630	4,442	3,285
ers)		Specific Range/Altitude	0.243/FL 470	0.146/FL 450	0.228/FL 470	0.135/FL 490	0.183/FL 470
		Runway	3,050	3,304	2,880	2,693	2,640
	1,000 nm	Flight Time	2+21	2+18	2+20	2+13	2+17
	1,000 mm	Fuel Used	3,755	6,176	4,075	6,747	4,945
		Specific Range/Altitude	0.266/FL 470	0.162/FL 450	0.245/FL 470	0.148/FL 470	0.202/FL 470
			FAR/EASA 25, 2013		FAR 25/EASA 25,	FAR 25, 1998/2004	FAR/EASA 25, 2007
emarks		Certification Basis	EASy II flight deck;	FAR 25, 2004	1979/2010 EASy II flight	and EASA 25, 2004	EASy II flight deck; DFC
			2015 delivery price.		deck; 2015 delivery price.	Global Vision flight deck.	2015 delivery price

### JETS 20,000-LB. MTOW OR GREATER

Manufacturer Model			Embraer Lineage 1000E ERJ 190-100 ECJ	Airbus ACJ318 A318-112	Boeing BBJ2 737-800	Airbus A320 Prestige A320-214	Boeing BBJ3 737-900ER
B&CA Equipped Price			\$53,000,000	\$72,000,000	\$91,500,000	\$95,000,000	\$99,300,000
		Seating	3+13/19	4+18/132	4+19/189	4+18/179	4+19/215
Character-		Wing Loading	120.7	113.6	129.9	130.3	139.9
stics		Power Loading	3.25	3.22	3.19	3.18	3.44
	Noise	(EPNdB): TO/Sideline/APR	86.4/92.7/92.5	83.0/91.9/93.9	86.0/94.4/96.4	85.5/93.4/95.5	88.4/93.8/96.4
xternal		Length	118.9	103.2	129.5	123.3	138.2
imensions		Height	34.7	41.1	41.2	38.6	41.2
t.)		Span	94.2	111.8	117.4	111.8	117.4
iternal		Length: OA/Net	84.3/84.3	70.2/70.2	98.3/98.3	90.3/90.3	107.2/107.2
imensions		Height	6.6	7.4	7.1	7.4	7.1
t.)		Width: Max/Floor	8.8/8.0	12.1/11.7	11.6/10.7	12.1/11.7	11.6/10.7
		Internal: Cu. ft./lb.	323/2,293	395/NA	NA/NA	NA/NA	NA/NA
aggage		External: Cu. ft./lb.	120/705	430/NA	721/NA	985/NA	882/NA
		External. ou. rt./ ib.	2 GE	2 CFMI	2 CFMI	2 CFMI	2 CFMI
		Engines	CF34-10E7-B	CFM56-5B9/3*	CFM56-7B27E	CFM56-5B4/3*	CFM56-7B27E
ower	0	utput (lb. each)/Flat Rating	18,500/ISA+15C	23,300/ISA+30C	27,300/ISA+15C	27,000/ISA+29C	27,300/ISA+15C
		Inspection Interval	OC	OC	OC	OC	OC
		Max Ramp	120,593	150,800	174,700	172,850	188,200
		Max Takeoff	120,152	149,900	174,200	171,950	187,700
		Max Landing	100,972	126,765	146,300	145,500	157,300
		Zero Fuel	80,469c	120,150c	138,300c	137,800c	149,300c
		BOW	70,548	96,694	103,800	109,000	111,650
		Max Payload	9,921	23.456	34,500	28,800	37,650
eights (lb.)		Useful Load	50,045	54,106	70,900	63,850	76,550
		Executive Payload	2,600	3,600	3,800	3,600	3,800
		Max Fuel	48,217	48,660	69,961	53,450	73,472
	Ava	ailable Payload w/Max Fuel	1,828	5,446	939	10,400	3,078
		ailable Fuel w/Max Payload	40,124	30,650	36,400	35,050	38,900
		e Fuel w/Executive Payload	47,445	48,660	67,100	53,450	72,750
		MMO	0.820	0.820	0.820	0.820	0.820
mits		Trans. Alt. FL/VMo	FL 289/320	FL 250/350	FL 260/340	FL 250/350	FL 260/340
		PSI	8.8	8.2	9.0	8.3	9.0
		TOFL (SL elev./ISA temp.)	6,076	5,870	6,670	6,920	8,350
		TOFL (5,000' elev.@25C)	9,500	7,660	12,850	9,355	14,500
rport		Hot/High Weight Limit	112,038	149,900	174,200	171,950	175,500p
erfor-		NBAA IFR Range	3,965	4,250	5,648	4,300	4,751
ance		V2@SLISA, MTOW	140	NA	152	NA	NA
ance	VREF W/4 Pax, NBAA IFR Res.		110	NA	121	NA	122
	Landing Distance w/4 Pax, NBAA IFR Res.		2,038	2,150	2,370	2,400	2,510
	Time to Climb/Altitude		29/FL 350	20/FL 370	27/FL 370	23/FL 360	26/FL 350
limb	FAR 25 Engine-Out Rate (fpm)		NA	NA	NA	NA	NA
	FAR 25 Engine-Out Gradient (ft./nm)		NA	NA	NA	NA	NA
	Certificated		41,000	41,000	41,000	39,000	41,000
	All-Engine Service		35,000	NA	37,700	NA	35,000
eilings (ft.)		Engine-Out Service	19,178	NA	20,000	NA	NA
	Sea-Level Cabin		23,190	MA	24,000	NA	24,000
		TAS	454	444	454	451	455
	Long	Fuel Flow	4,184	4,230	5,043	4,730	5,427
	Range	Altitude	FL 380	FL 390	FL 390	FL 370	FL 360
	mingo	Specific Range	0.109	0.105	0.090	0.095	0.084
ruise		TAS	471	470	470	473	473
	High	Fuel Flow	5,033	5,360	5,721	5,860	6,272
	Speed	Altitude	FL 350	FL 370	FL 360	350	FL 350
	-opoou	Specific Range	0.094	0.088	0.082	0.081	0.075
		Nautical Miles	3,493	2,048	2,279	2,100	2,223
	Max Payload	Average Speed	442	426	433	428	436
	(w/available fuel)		35,569	24,129	29,968	27,936	
		Irip Fuel					32,052
		Specific Range/Altitude	0.098/FL 400	0.085/FL 370	0.076/FL 370	0.075/FL 350	0.069/FL 350
		Nautical Miles	4,532	4,000	5,726	3,852	5,452
BAA IFR		Average Speed	446	437	445	438	447
		Trip Fuel	43,962	42,710	64,835	46,930	67,949
anges		Specific Range/Altitude	0.103/FL 410	0.094/FL 410	0.088/FL 410	0.082/FL 390	0.080/FL 390
)0-nm		Nautical Miles	4,602	4,300	5,622	4,330	5,496
ernate)		Average Speed	446	436	444	438	446
		Trip Fuel	44,240	43,601	63,899	48,057	67,985
		Specific Range/Altitude	0.104/FL 410	0.099/FL 410	0.088/FL 410	0.090/FL 390	0.081/FL 390
		Nautical Miles	4,640	4,300	5,754	4,380	5,555
		Average Speed	446	436	444	438	447
		Trip Fuel	44,264	43,653	64,855	48,108	68,030
		Specific Range/Altitude	0.105/FL 410	0.099/FL 410	0.089/FL 410	0.091/FL 390	0.082/FL 390
		Runway	3,002	3,675	3,245	3,670	3,700
	300 nm	Flight Time	0+48	0+53	0+56	0+55	0+55
	0001111	Fuel Used	3,426	4,077	4,547	4,265	4,841
		Specific Range/Altitude	0.088/FL 390	0.074/FL 370	0.066/FL 310	0.070/FL 350	0.062/FL 290
ssions		Runway	3,133	3,700	3,365	3,700	3,855
passen-	600 nm	Flight Time	1+26	1+33	1+32	1+34	1+32
	000 mm	Fuel Used	5,862	6,694	7,268	7,080	7,775
rs)		Specific Range/Altitude	0.102/FL 410	0.090/FL 410	0.083/FL 410	0.085/FL 390	0.077/FL 390
		Runway	3,251	3,760	3,535	3,760	4,045
	1 000 pm	Flight Time	2+20	2+27	2+26	2+28	2+26
	1,000 nm	Fuel Used	9,063	10,225	11,088	10,970	11,943
		Specific Range/Altitude	0.110/FL 410	0.098/FL 410	0.090/FL 410	0.091/FL 390	0.084/FL 390
emarks		Certification Basis	FAR/EASA 25, 2008	FAR 25, 2003 *Also avail with PW6124; incl. 2 add'l center tanks; price incl. VIP cabin.	FAR 25 A 77, 1967/98 All pax and range missions flown with eight passengers.	FAR 25, 1999 *Also avail. with 26,500-lbf IAEV2527M-A5; incl. 2 add'l center tanks and VIP cabin.	FAR 25 A 77, 1967/

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### **ULTRA-LONG-RANGE JETS**

Manufacturer			Gulfstream Aerospace	Bombardier	Gulfstream Aerospace	Boeing	Airbus	
Model			Gulfstream 550 GV-SP	Global 6000 BD-700-1A10	Gulfstream 650 GVI	BBJ 737-700IGW	ACJ319 A319-133	
3&CA Equippe	d Price		\$60,000,000	\$62,000,000	\$65,200,000*	\$73,500,000	A319-133 \$87,000,000	
Con Equippe		Seating	4+16/19	4+13/19	4+16/19	4+19/149	4+19/156	
Character-		Wing Loading	80.1	97.4	77.6	127.5	127.8	
stics		Power Loading	2.96	3.37	2.95	3.13	3.12	
	Noise	(EPNdB): TO/Sideline/APR	79.3/90.2/90.8	82.4/87.9/89.7	77.5/89.8/88.3	85.4/94.9/95.8	85.4/94.6/94.2	
xternal		Length	96.4	99.4	99.8	110.3	111.0	
imensions		Height	25.8	25.5	25.7	41.2	38.6	
t.)		Span	93.5	94.0	99.6	117.4	111.8	
iternal		Length: OA/Net	50.1/42.6	48.4/43.2	53.6/46.8	79.2/79.2	78.0/78.0	
imensions		Height	6.2	6.2	6.4	7.1	7.4	
t.)		Width: Max/Floor	7.3/5.5	7.9/6.5	8.5/7.0	11.6/10.7	12.2/11.6	
		Internal: Cu. ft./lb.	226/2,500	195/1,000	235/2,500	NA/NA	160/NA	
aggage		External: Cu. ft./lb.	_/_	_/_	_/_	169/NA	NA/NA	
			2 RR	2 RR	2 RR	2 CFMI	2 CFMI	
		Engines	BR700-710C4-11	BR700-710A2-20	BR700-725A1-12	CFM56-7B27E	CFM56-5B7/3*	
ower	01	utput (lb. each)/Flat Rating	15,385/ISA+15C	14,750/ISA+20C	16,900/ISA+15C	27,300/ISA+15C	27,000/ISA+29C	
		Inspection Interval	8,000t or 0C	00	10,000t	00	00	
		Max Ramp	91,400	99,750	100,000	171,500	169,530	
		Max Takeoff	91,000	99,500	99,600	171,000	168,650	
		Max Landing	75,300	78,600	83,500	134,000	137,790	
		Zero Fuel	54,500c	58,000c	60,500c	126,000c	128,970c	
		BOW . Max Payload	48,700 5,800	52,230 5,770	54,000 6,500	97,740 28,260	96,450** 32,520	
eights (lb.)		Useful Load	42,700	47,520	46,000	73,760	73,080	
		Executive Payload	3,200	2,600	3,200	3,800	3,800	
		Max Fuel	40,994	44,716	44,200	71,737	72,560	
	Ava	ailable Payload w/Max Fuel	1,706	2,804	1,800	2,023	520	
	Ava	ailable Fuel w/Max Payload	36,900	41,750	39,500	45,500	40,560	
	Available	e Fuel w/Executive Payload	39,500	44,716	42,800	69,960	69,280	
1		Ммо	0.885	0.890	0.925	0.820	0.820	
mits		Trans. Alt. FL/VMO	FL 270/340	FL 303/340	FL 290/340	FL 260/340	FL 250/350	
		PSI	10.2	10.3	10.7	9.0	8.3	
		TOFL (SL elev./ISA temp.)	5,910	6,476	5,858	6,085	6,170	
		TOFL (5,000' elev.@25C) Hot/High Weight Limit	9,070 91.000	7,880 94,513p	9,000 99,600	10,200 170,827	8,360 168,650	
rport		NBAA IFR Range	6,738	5,633	7,000	6,260	6,000	
erformance	V2 @ SL, ISA, MTOW VREF w/4 Pax, NBAA IFR Res.		147	142	146	141	137	
			112	110	114	116	111	
		ce w/4 Pax, NBAA IFR Res.	2,240	2,236	2,667	2,360	2,220	
	Time to Climb/Altitude FAR 25 Engine-Out Rate (fpm)		18/FL 370	20/FL 370	20/FL 370	25/FL 370	22/360	
limb			594	474	NA	NA	NA	
	FAR 25 Engine-Out Gradient (ft./nm)		242	200	NA	NA	NA	
	Certificated		51,000	51,000	51,000	41,000	41,000	
eiling (ft.)	All-Engine Service Engine-Out Service		42,700	42,400	42,700	38,000	36,000	
8 ()			25,820	18,000	25,000	20,500	18,000	
	Sea-Level Cabin TAS		29,200 459	30,125 470	31,900 488	24,000 452	22,000 447	
	Long		2,563	3,043	2,686	4.707	447	
	Range	Fuel Flow _ Altitude	FL 450	FL 450	FL 470	FL 390	FL 370	
	Rungo	Specific Range	0.179	0.154	0.182	0.096	0.095	
Cruise		TAS	488	499	516	470	470	
		Fuel Flow	3,228	3,789	3,368	5,567	5,830	
	High Speed	Altitude	FL 430	FL 410	FL 430	FL 370	FL 370	
		Specific Range	0.151	0.132	0.153	0.084	0.081	
		Nautical Miles	5,767	5,876	5,980	3,291	2,679	
	Max Payload	Average Speed	452	463	474	437	434	
	(w/available fuel)	Trip Fuel	33,993	40,480	36,500	39,571	33,677	
		Specific Range/Altitude	0.170/FL 490	0.145/FL 450	0.164/FL 490	0.083/FL 390	0.080/FL 370	
	Max Fuel	Nautical Miles	6,698	6,111	7,000	6,229	6,134	
BAA IFR	(w/available	Average Speed	454	464	476	442	442	
	payload)	Trip Fuel	38,202	41,780	41,200	66,866	66,673	
anges 00-nm		Specific Range/Altitude Nautical Miles	0.175/FL 490 6,708	0.146/FL 450	0.170/FL 510 7,000	0.093/FL 410 6,237	0.092/FL 410 6,002	
00-nm	Eight Passengers	Average Speed	453	6,163 464	476	442	6,002	
ternate)	(w/available fuel)	Average Speed Trip Fuel	38,205	404 41,780	41,200	66,871	65,558	
		Specific Range/Altitude	0.176/FL 490	0.148/FL 450	0.170/FL 510	0.093/FL 410	0.092/FL 410	
		Nautical Miles	6,853	6,258	7,157	6,306	6,200	
		Average Speed	454	464	477	442	442	
	Ferry	Trip Fuel	38,251	41,780	41,200	66,914	67,207	
		Specific Range/Altitude	0.179/FL 510	0.150/FL 470	0.174/FL 510	0.094/FL 410	0.092/FL 410	
		Runway	3,436	2,832	3,283	3,480	4,075	
	1,000 nm	Flight Time	2+20	2+13	2+12	2+27	2+26	
	1,000 mm	Fuel Used	5,599	6,838	5,891	10,422	10,370	
		Specific Range/Altitude	0.179/FL 490	0.146/FL 490	0.170/FL 510	0.096/FL 410	0.096/FL 410	
		Runway	3,599	3,818	3,586	4,275	4,280	
issions	3,000 nm	Flight Time	6+42	6+20	6+18	6+55	6+54	
passengers)		Fuel Used	15,474	19,461	16,191	29,650	30,070	
		Specific Range/Altitude	0.194/FL 490	0.154/FL 490	0.185/FL 510	0.101/FL 410	0.100/FL 410	
		Runway	5,277	6,144	5,214	5,870	6,160	
	6,000 nm	Flight Time	13+15	12+31	12+29	13+34	13+35	
		Fuel Used	33,428	41,780	34,313	63,852	65,528	
		Specific Range/Altitude	0.179/FL 490	0.144/FL 450	0.175/FL 510	0.094/FL 410	0.092/FL 410	
emarks		Certification Basis	FAR 25, 1997/2003/02	FAR 25, 1998/2003 and JAR 25 BEVS and Global Vision flight deck standard.	FAR, EASA CS 25, 2012 *B&CA estimate.	FAR 25 A 77, 1967/98	FAR 25, 1999 *Avail. w 26,500-lbf IAEV2527M- incl. 6 add'l center tank VIP cabin; **Spec wt	